

**Thermal diffusivity measurements as a non
destructive tool for the microstructural
characterisation and the integrity assessment of
thermal barrier coatings**

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Acknowledgments



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Summary

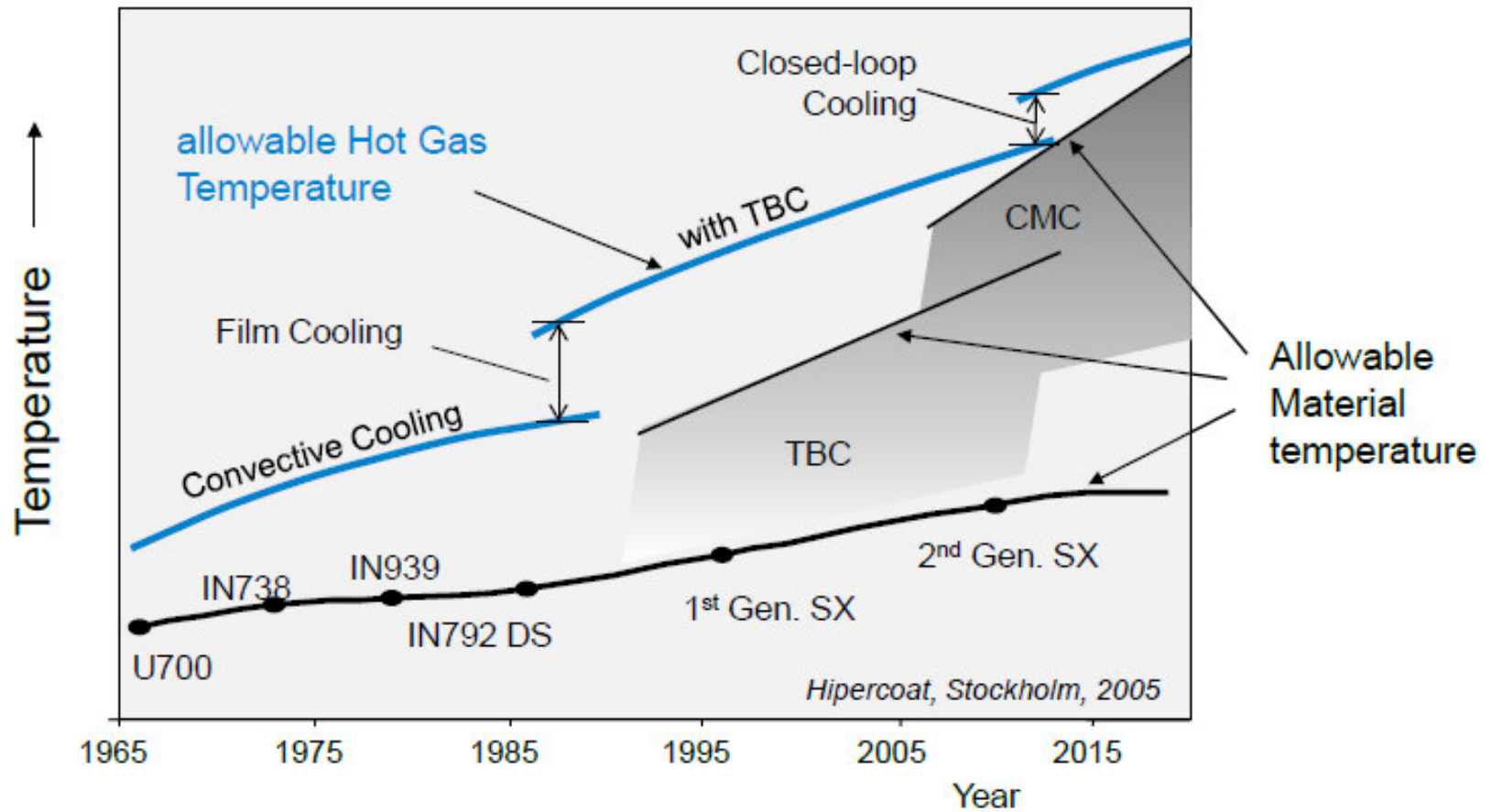


- Introduction to Thermal Barrier coatings (TBCs)
 - Integrity assessment of TBC by thermal diffusivity evaluation
 - Microstructural characterisation of TBC by thermal diffusivity evaluation and sintering forecasts
 - Through-the-thickness and in-plane thermal diffusivity measurements
-

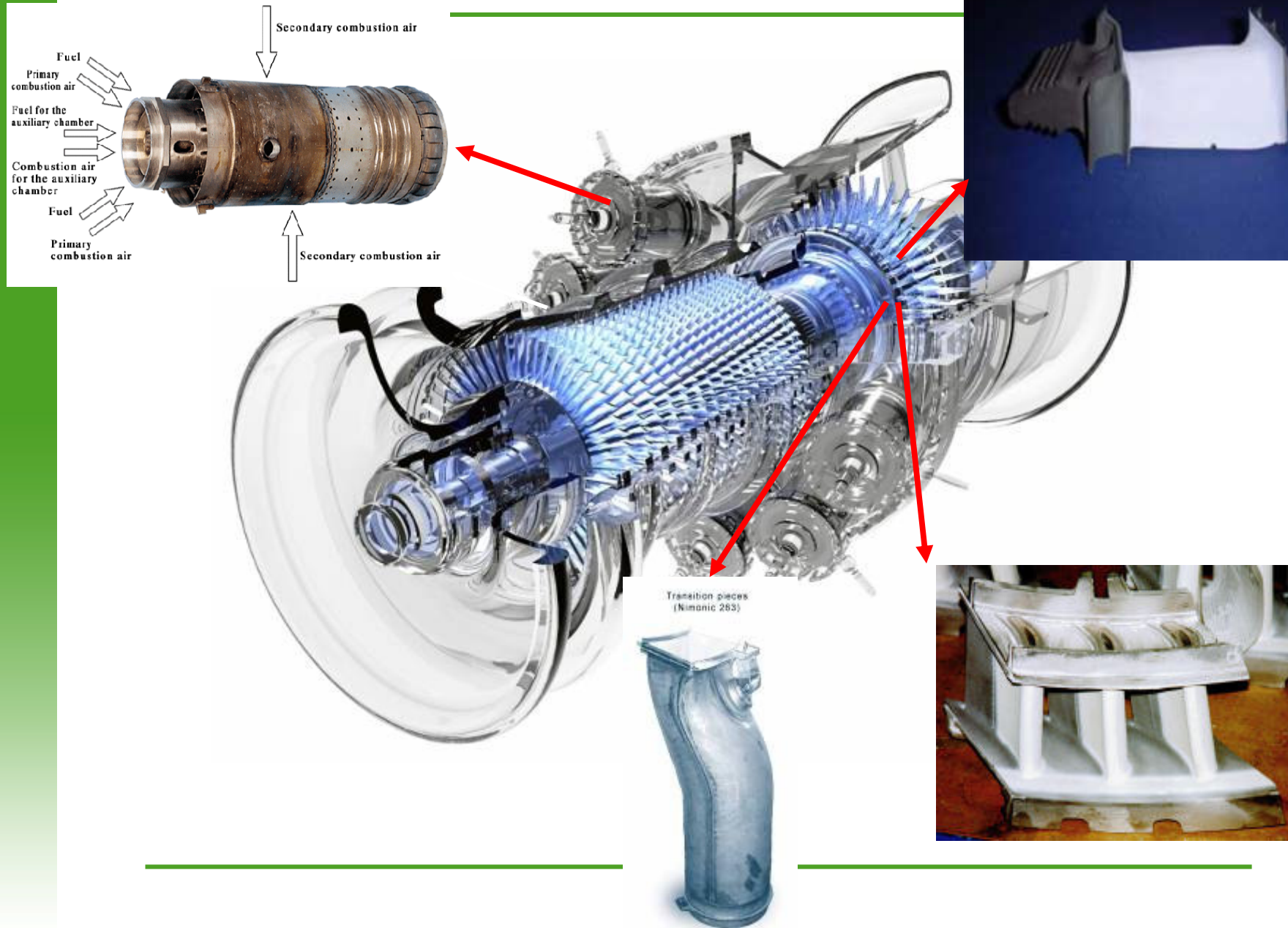
The gas turbine



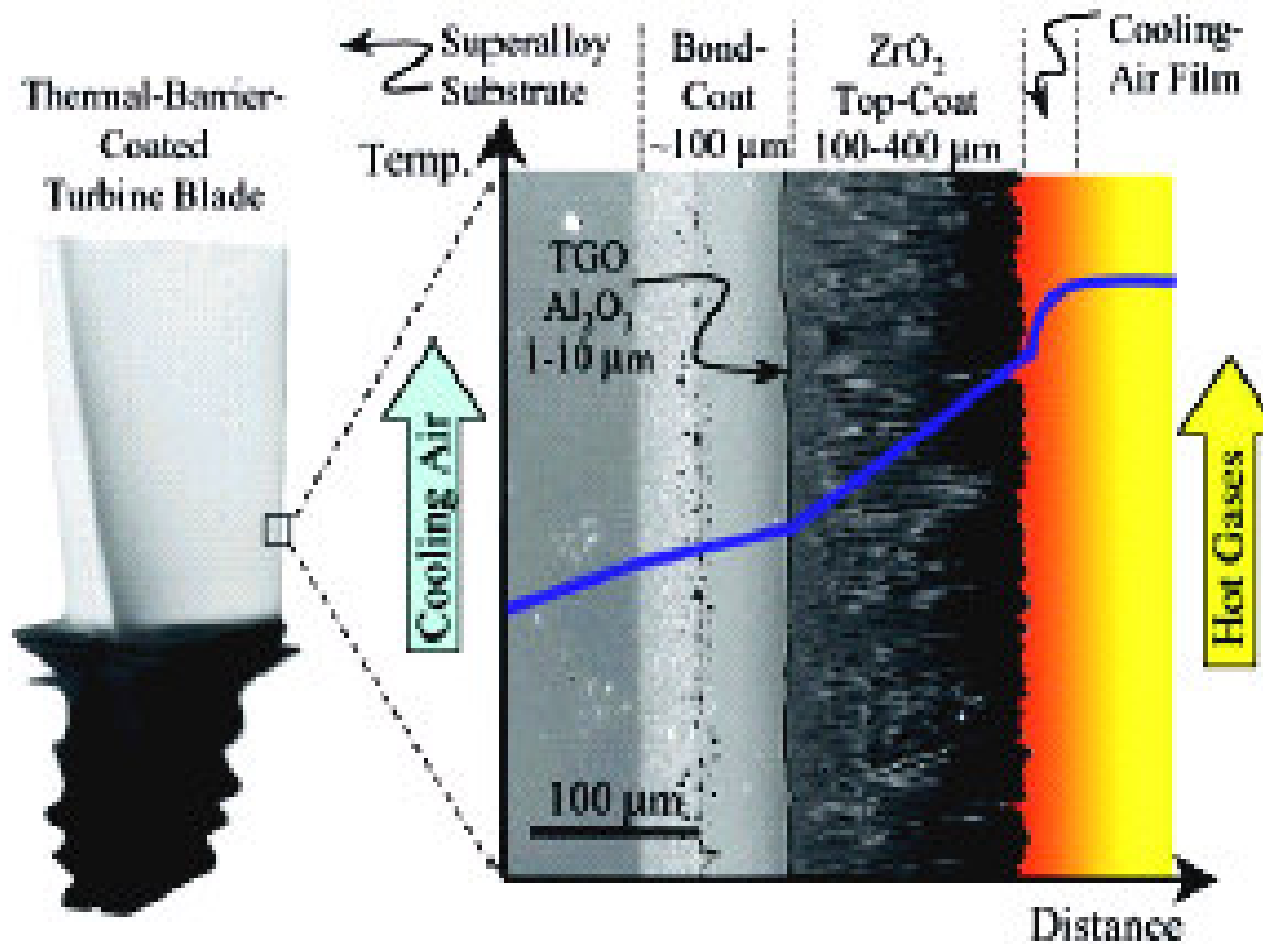
GT Inlet Temperature / GT Efficiency



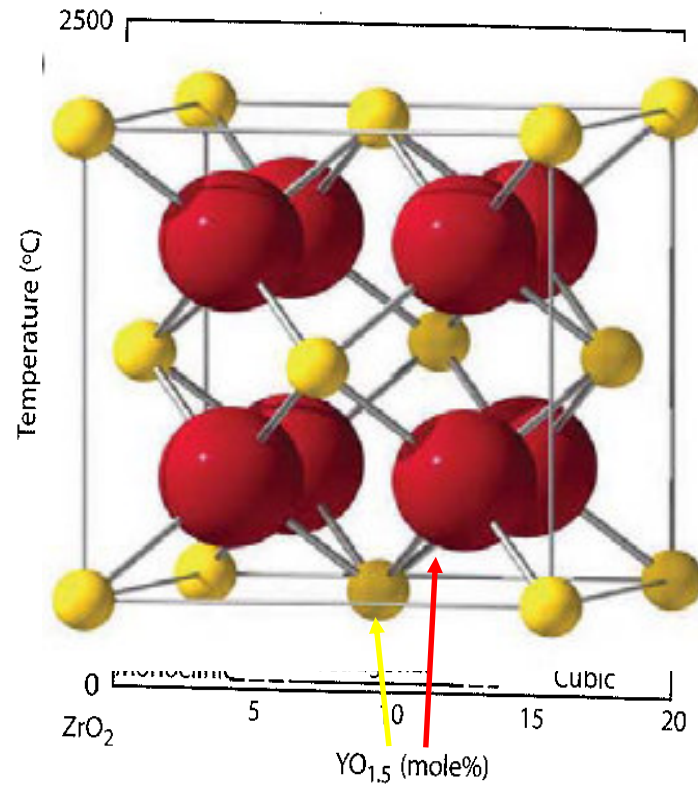
The GT hot path components



The thermal barrier coating (TBC)



The material $8\% \text{Y}_2\text{O}_3 + \text{ZrO}_2$



Fluoritic AO_2

A +4

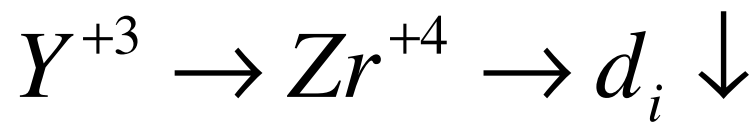
- **Maximum operating temperature $1200\text{ }^\circ\text{C}$**
- **TEC similar to that of substrate ($1 \cdot 10^{-5}$ vs. $1.2 - 1.3 \cdot 10^{-5}$)**
- **High toughness K_{1c} ($9.5 - 10.5 \text{ Mpa m}^{1/2}$)**
- **Low thermal conductivity ($2.8 - 2.2 \text{ W/mK}$)**
- **Coating $\sim 1 \text{ W/mK} \Leftrightarrow 50-170\text{ }^\circ\text{C}$**
- **Coating thickness $100-1000 \mu\text{m}$**

The material $8\% \text{Y}_2\text{O}_3 + \text{ZrO}_2$

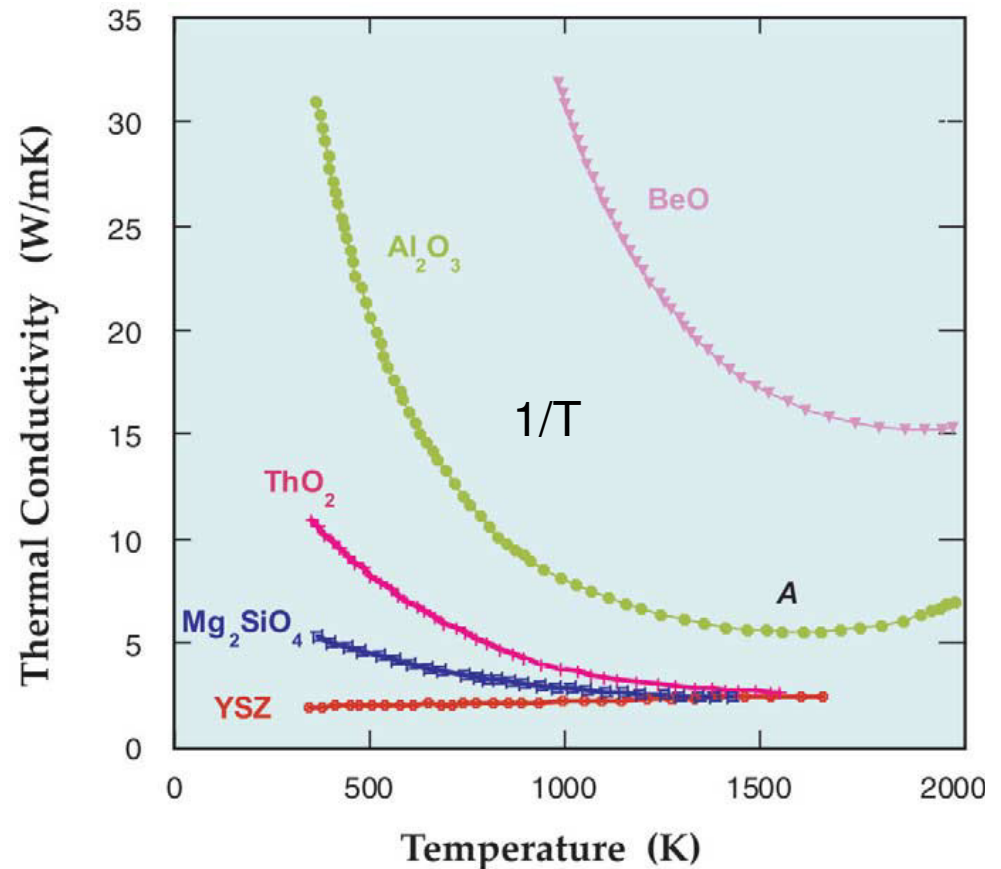


$$\kappa = \frac{1}{3} C_v v \Lambda$$

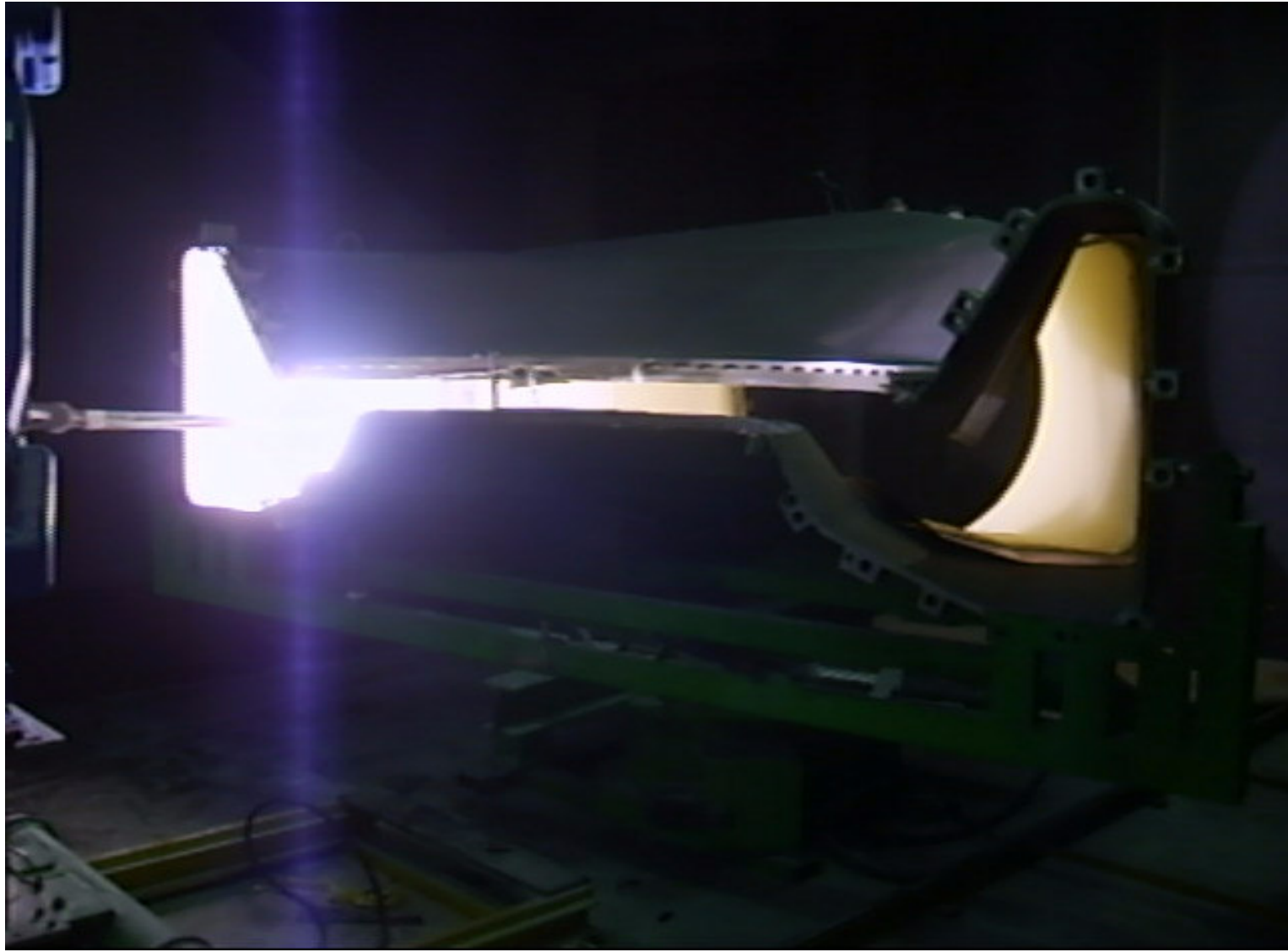
$$\Lambda \cong d_i$$



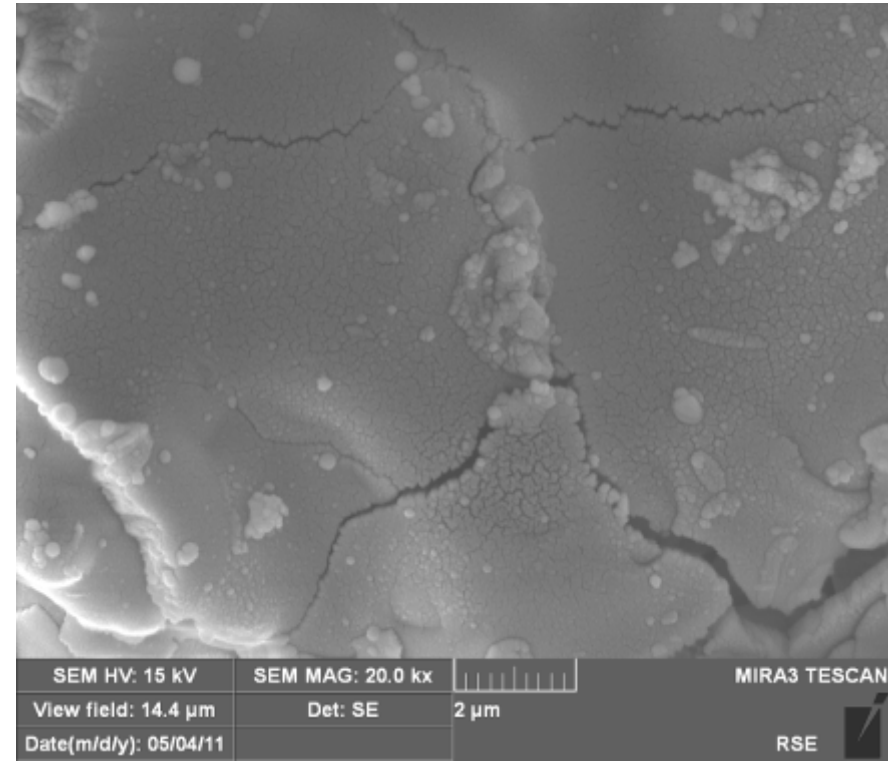
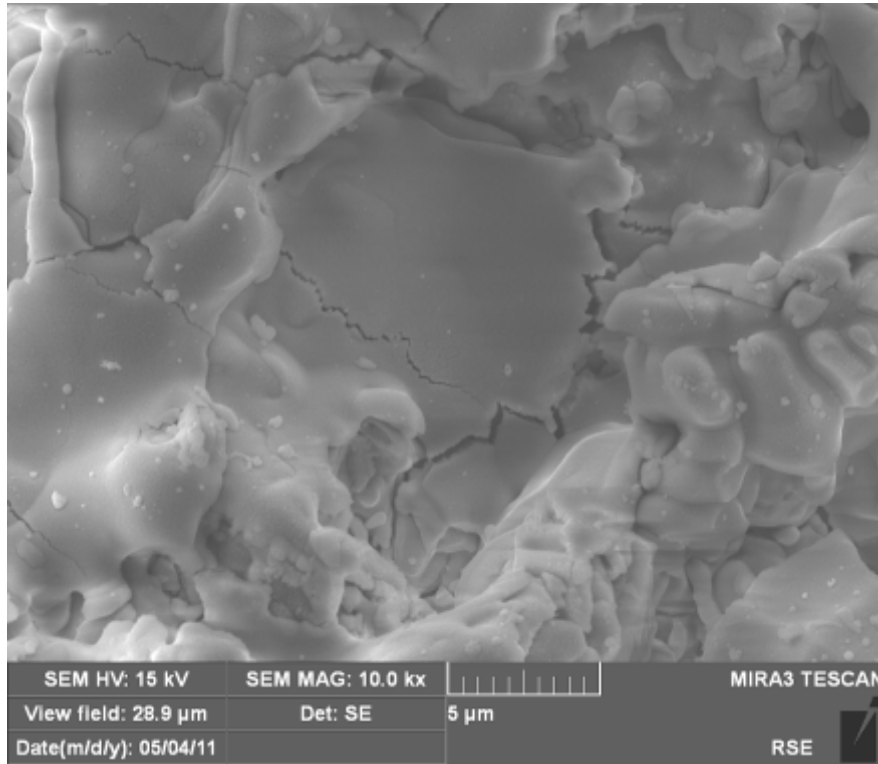
Non phonon vibrational modes with $v < v_{\text{sound}}$



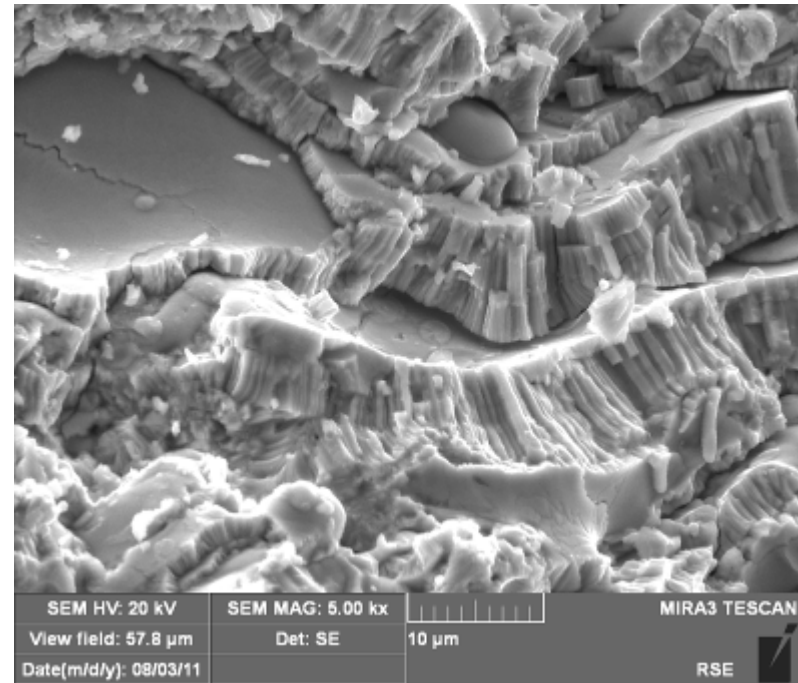
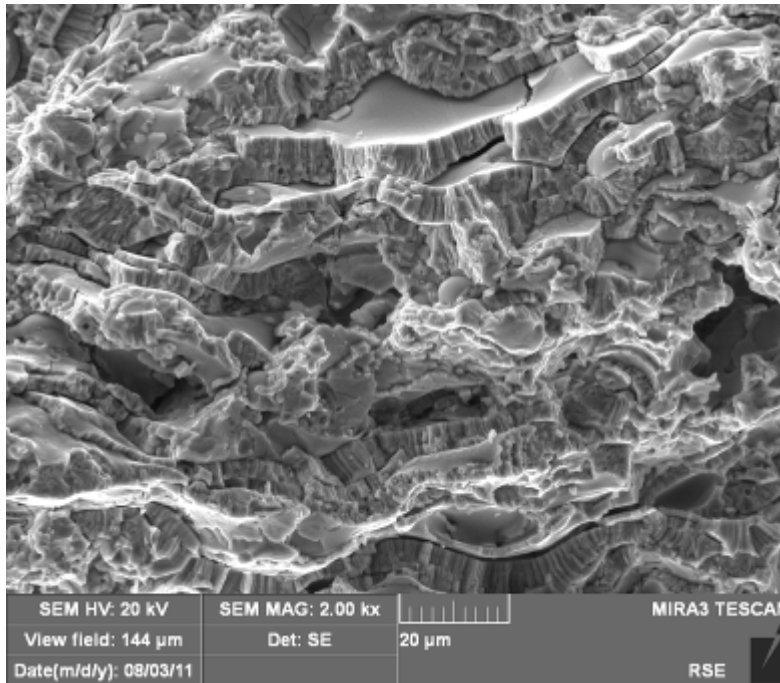
The deposition techniques: APS



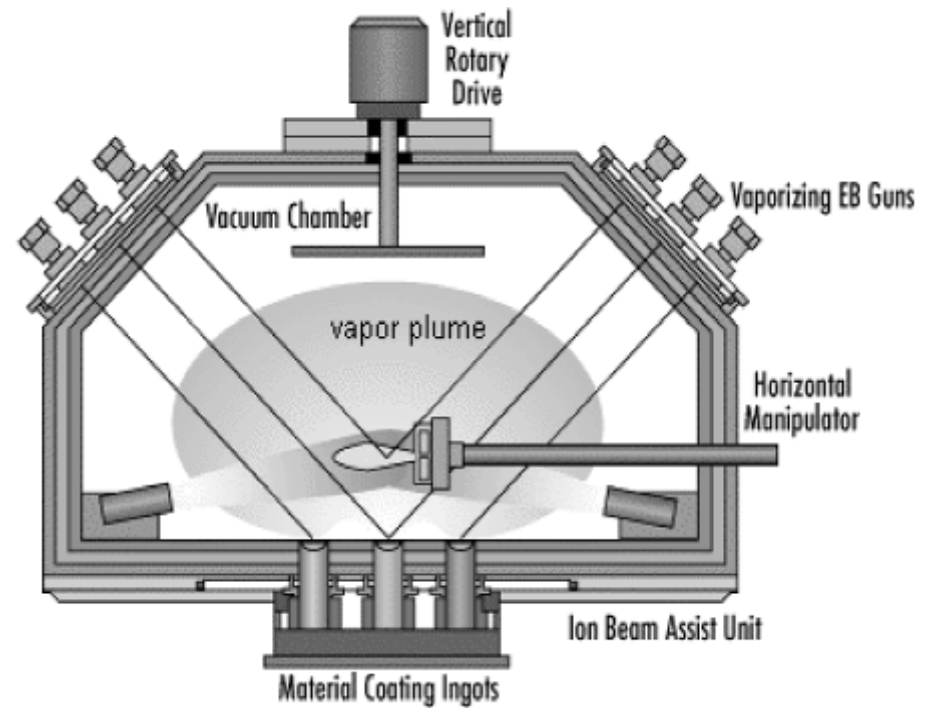
The deposition techniques: APS



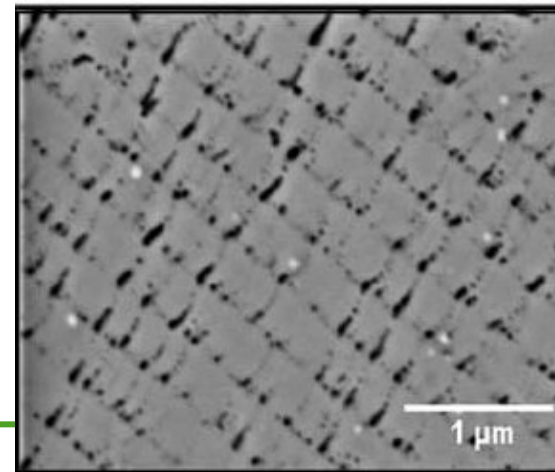
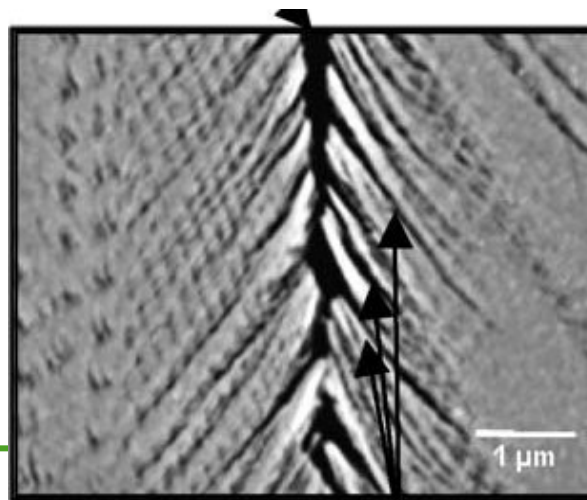
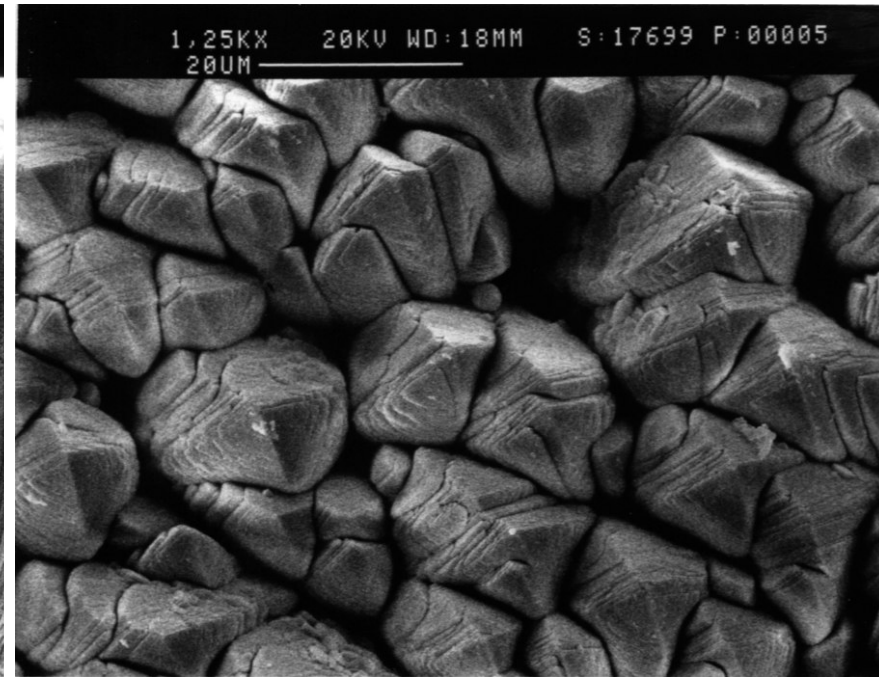
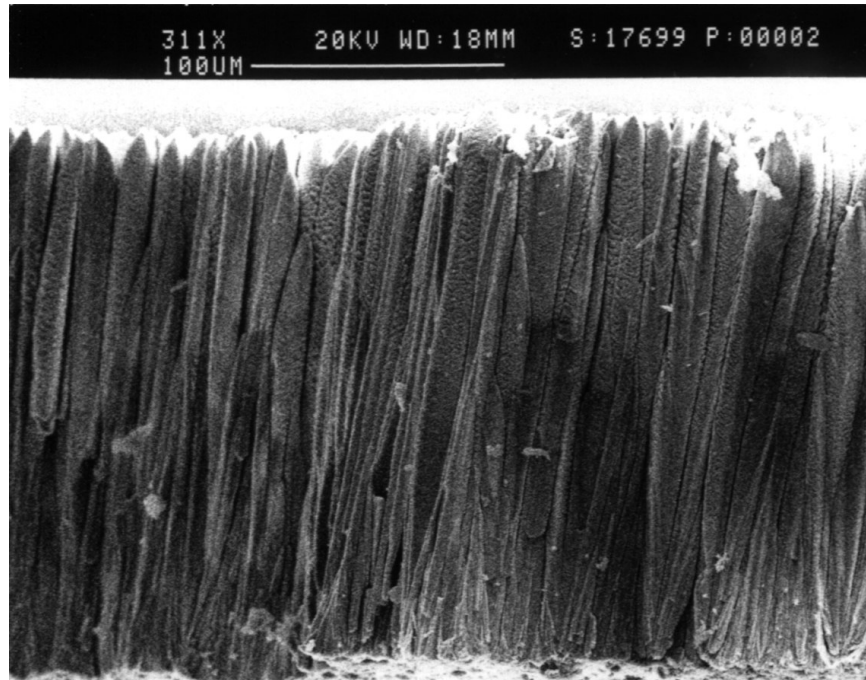
The deposition techniques: APS



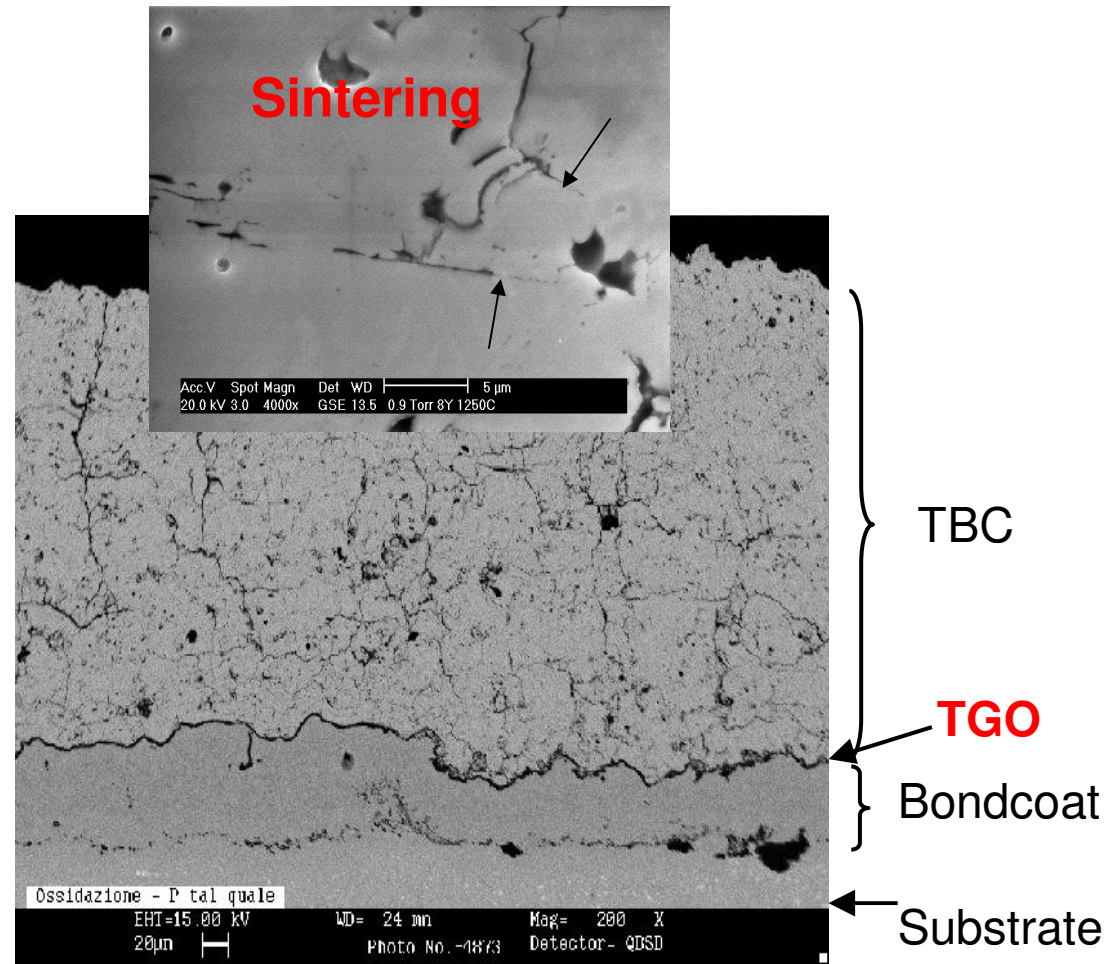
The deposition techniques: EBPVD



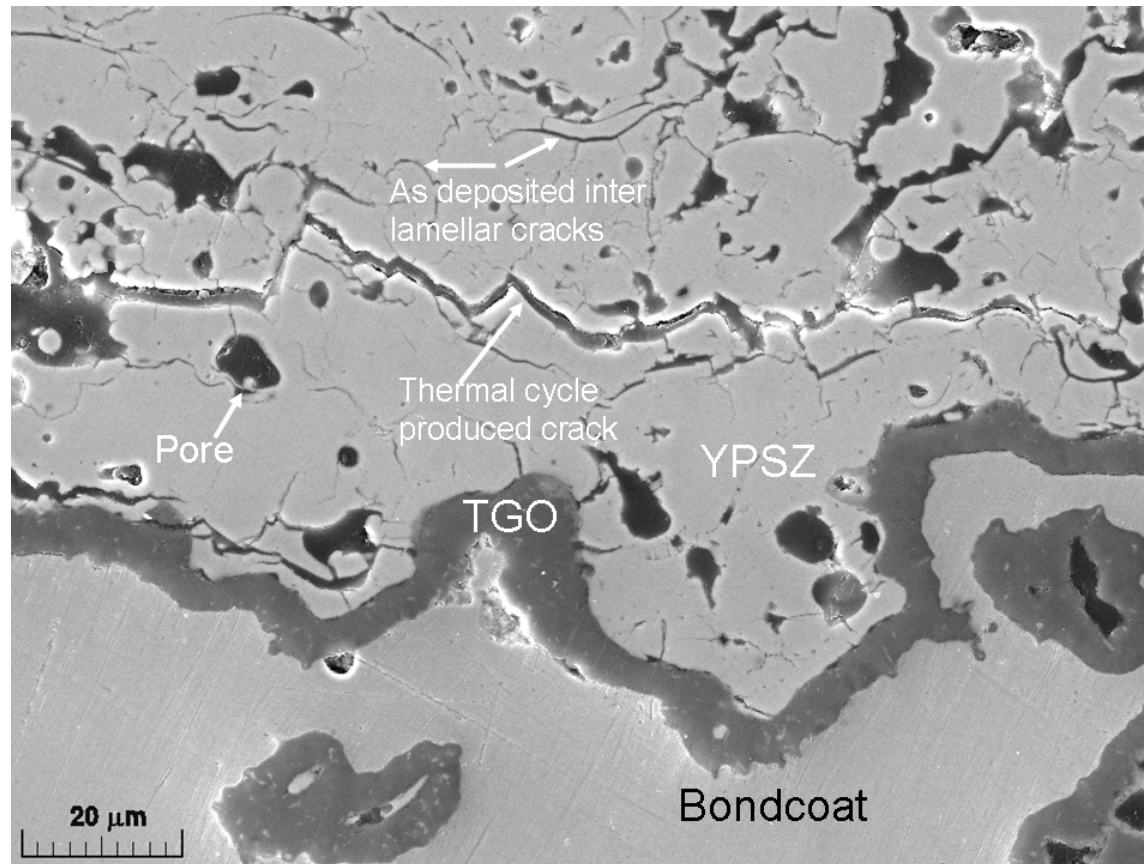
The deposition techniques: EBPVD



Failure mechanisms of TBC



Failure mechanisms of TBC



- **TGO growth**
- **Thermal fatigue**

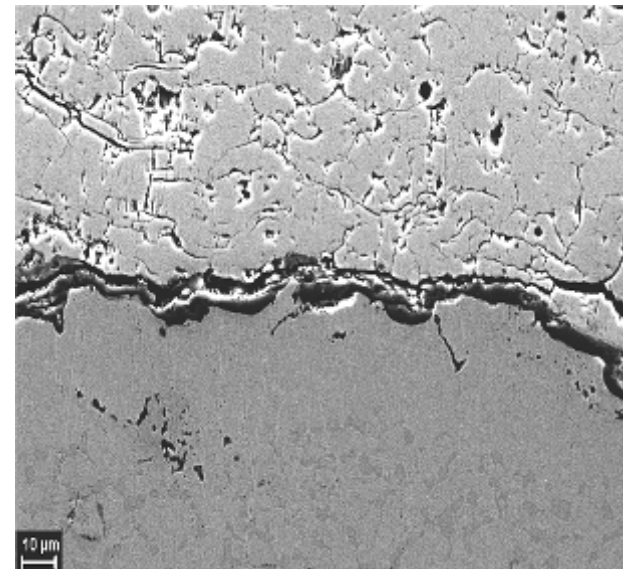
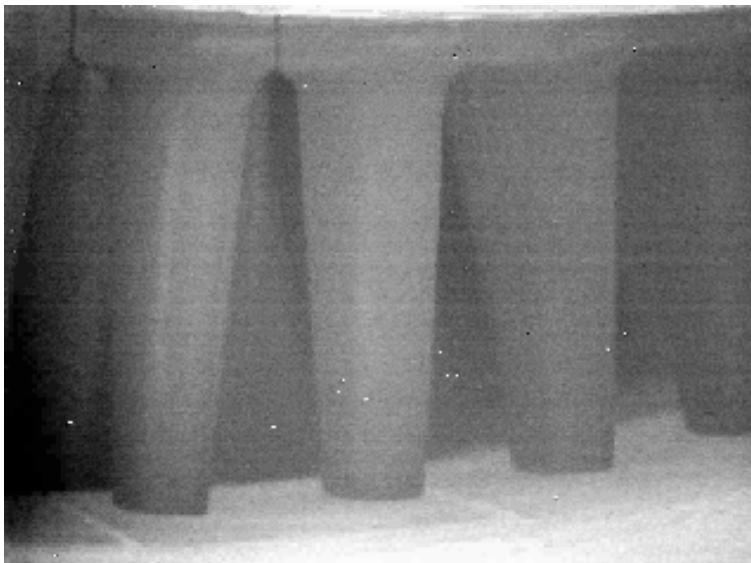
Non destructive integrity assessment of TBCs Coupons

F. Cernuschi, S. Capelli, P. Bison, S. Marinetti, L. Lorenzoni, E. Campagnoli and C. Giolli, *Acta Materialia* 59 (2011) 6351–6361

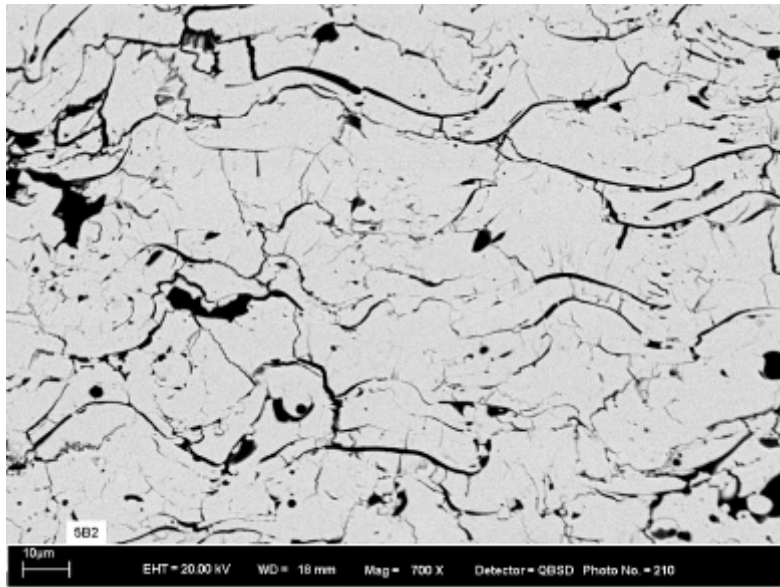
ND detection of **cracking** at interface BC – TBC



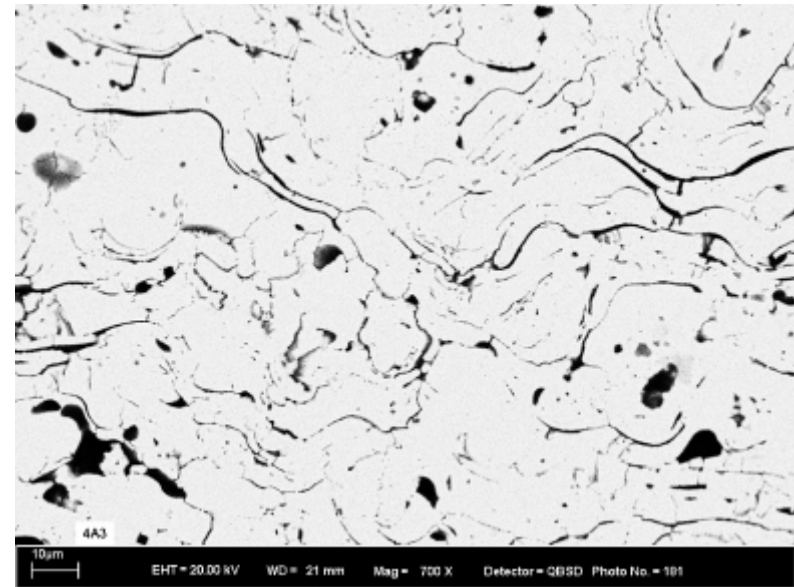
The objective is to non destructively detect **diffuse cracking** at the interface between BC and TBC in coupons tested under cyclic oxidation



Ageing of TBC:sintering



(T,t)

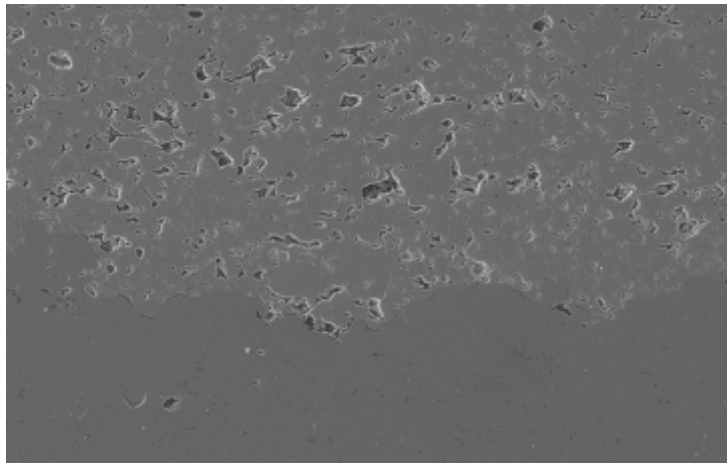


$(\alpha_0, E_0, H_0, \varphi_0, \dots)$

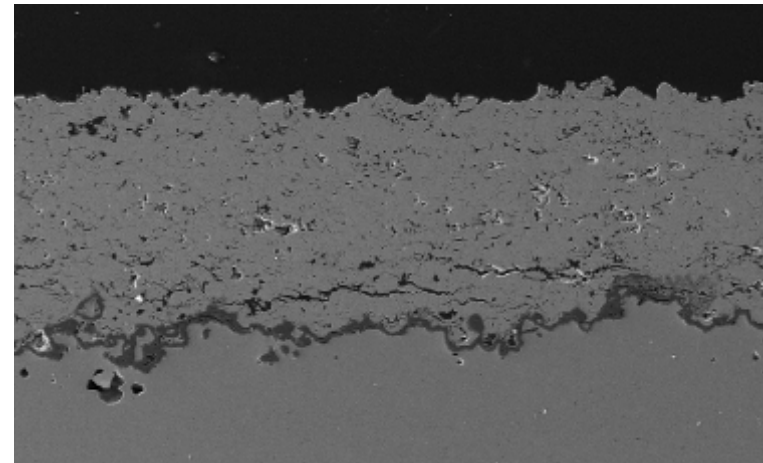


$(\alpha, E, H, \varphi \dots) > (\alpha_0, E_0, H_0, \dots)$

Ageing of TBC: cracking at the interface



(T, t)



(α_0, \dots)



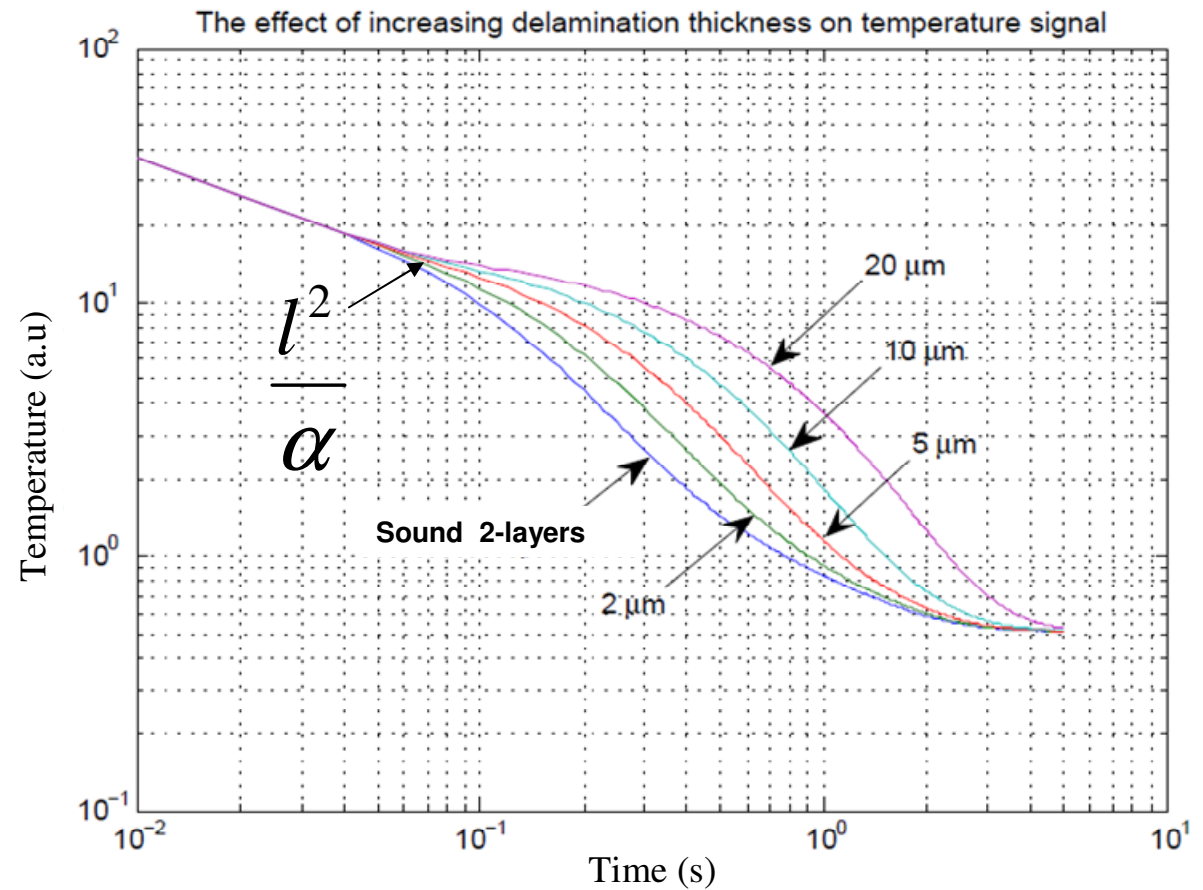
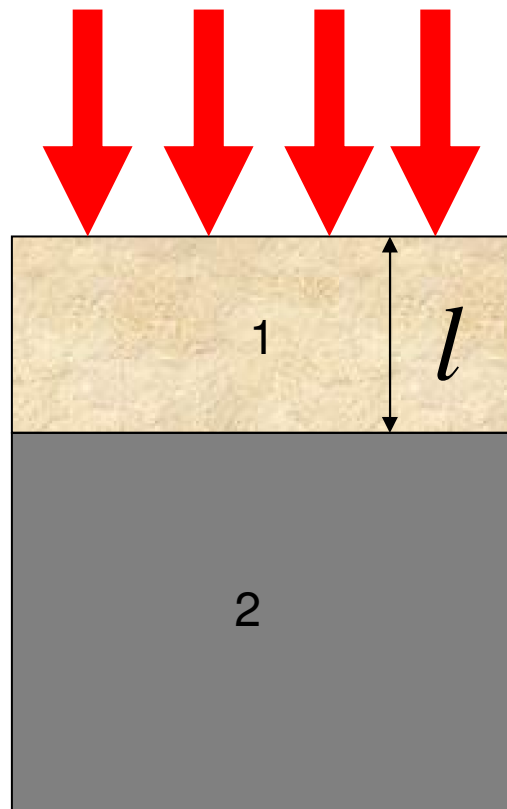
$(\alpha, \dots) < (\alpha_0, \dots)$



Integrity assessment: cracks and delaminations



Heat Dirac pulse



Crack growth ND detection on samples tested by cyclic oxidation



Superposition of two effects (sintering&cracking) with different relaxation times and magnitudes



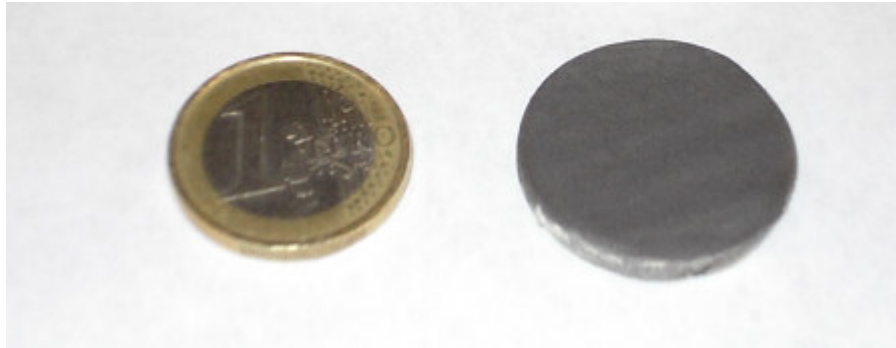
No monotonic trend of thermal diffusivity vs. ageing time



Some information about cracking at the interface can be obtained from thermal diffusivity



Crack growth ND detection on samples tested for cyclic oxidation



28 disk shaped samples coated with a APS TBC

9 thin samples (T1) $254 \pm 10 \mu\text{m}$

19 thick samples (T2) $328 \pm 9 \mu\text{m}$

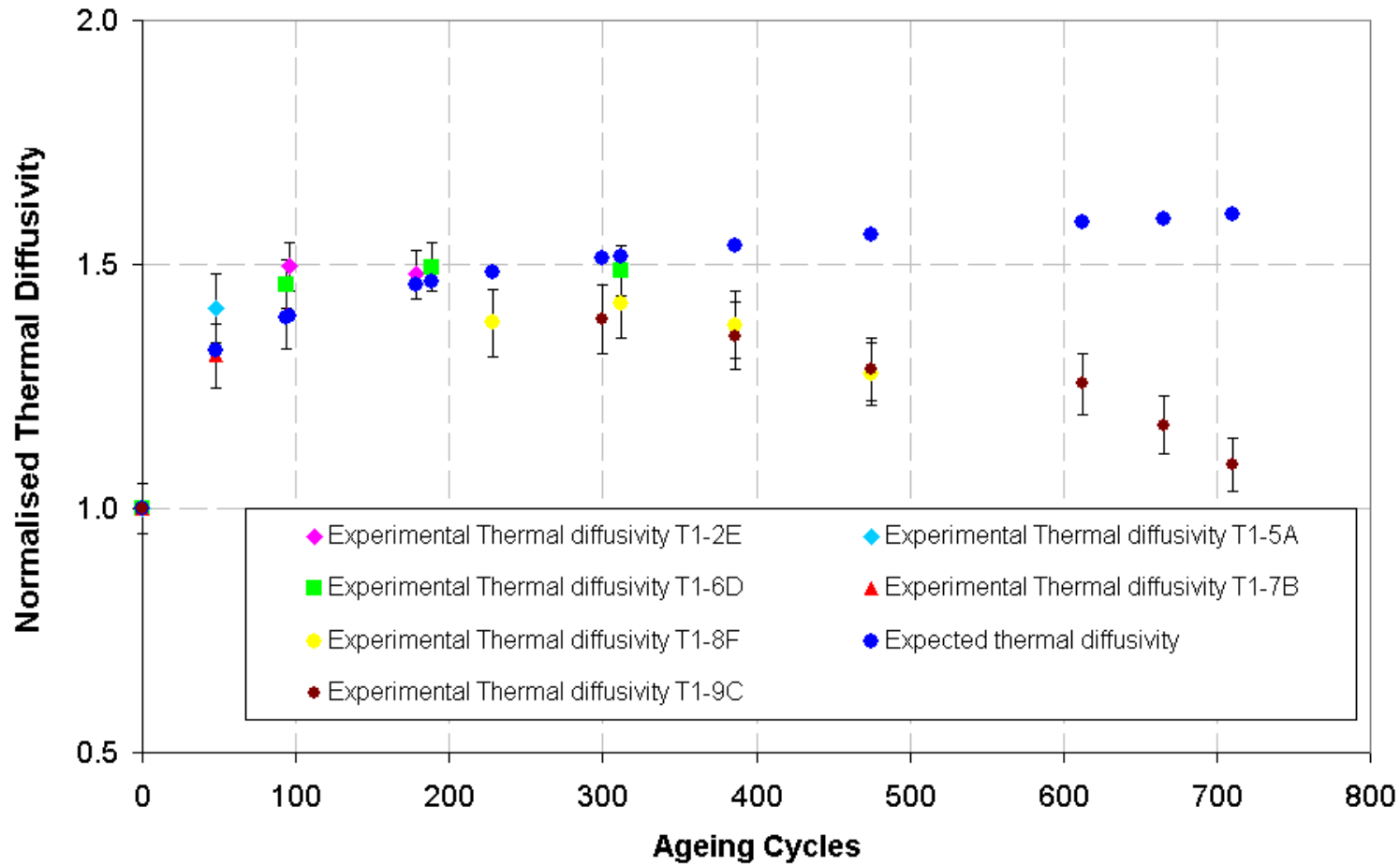
Porosity 15%

2 hours cycle: 1.5 hour dwell time @ 1050°C

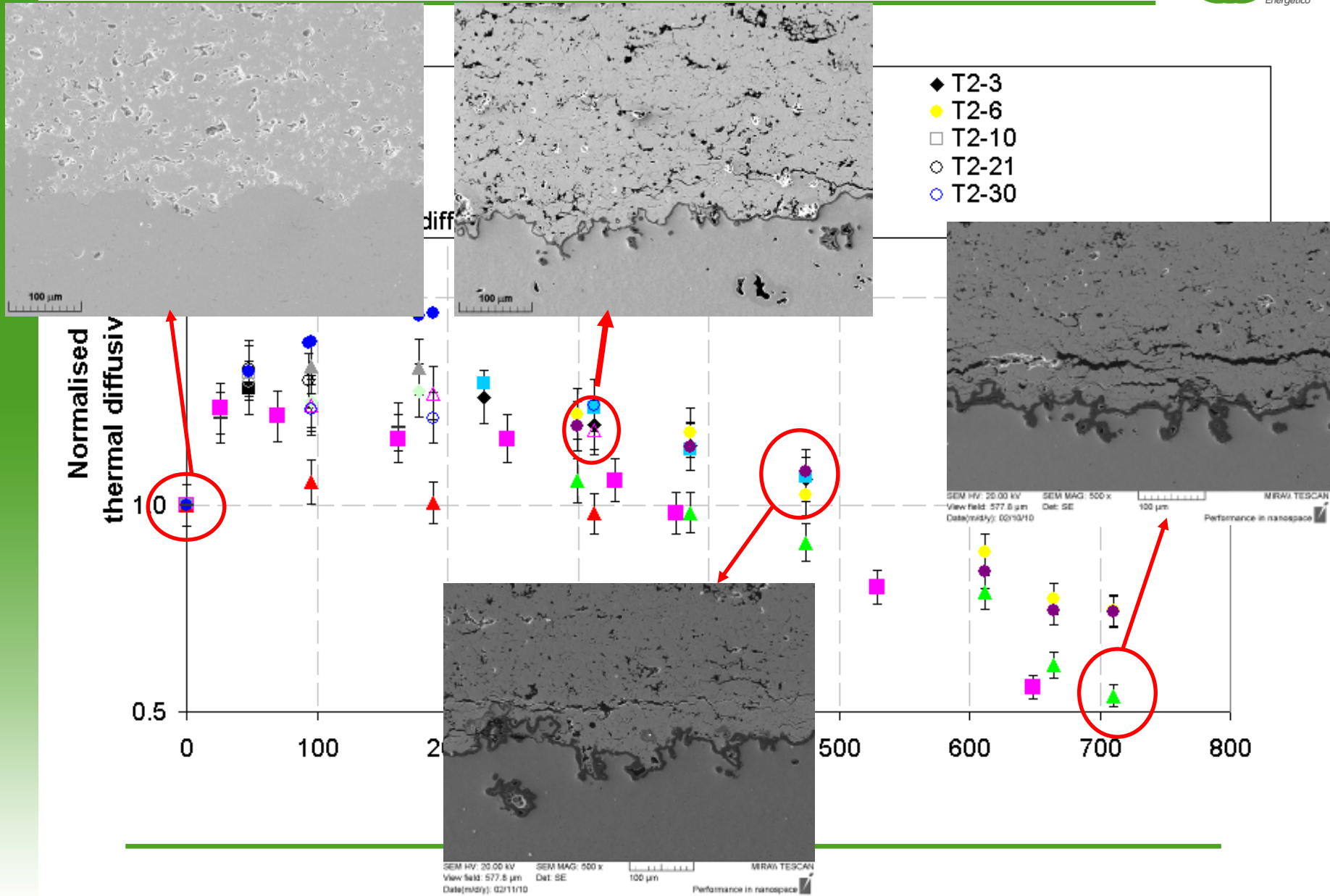
Average lifetime was estimated on 2+2 samples (T1=1008±12 and T2=880±12 cycles)

Samples have been aged up a fixed percentages of the average lifetime. Thermal diffusivity of each sample was measured in the as sprayed condition and after different aging cycles.

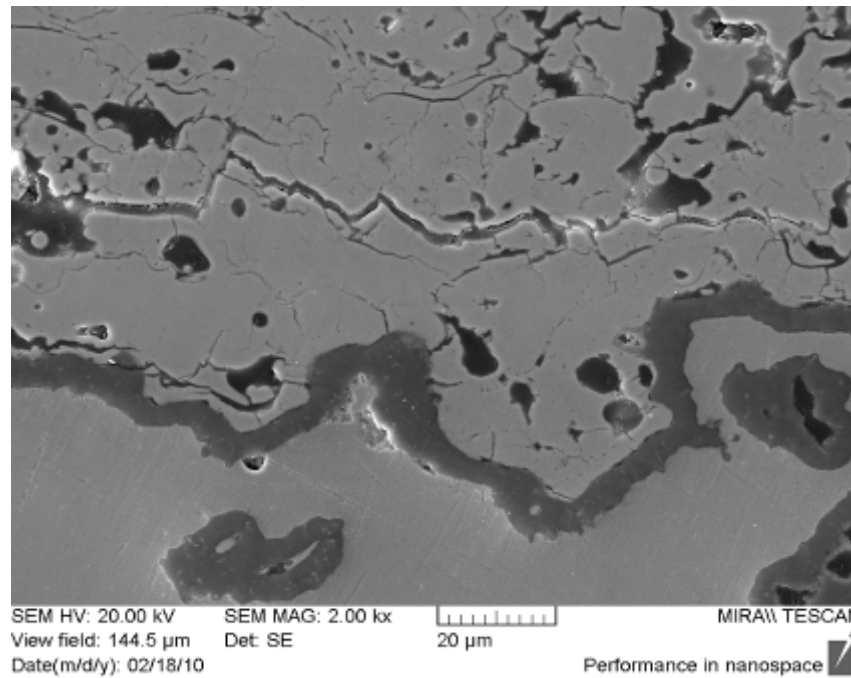
Crack growth ND detection on thermally cycled samples – Thin TBC

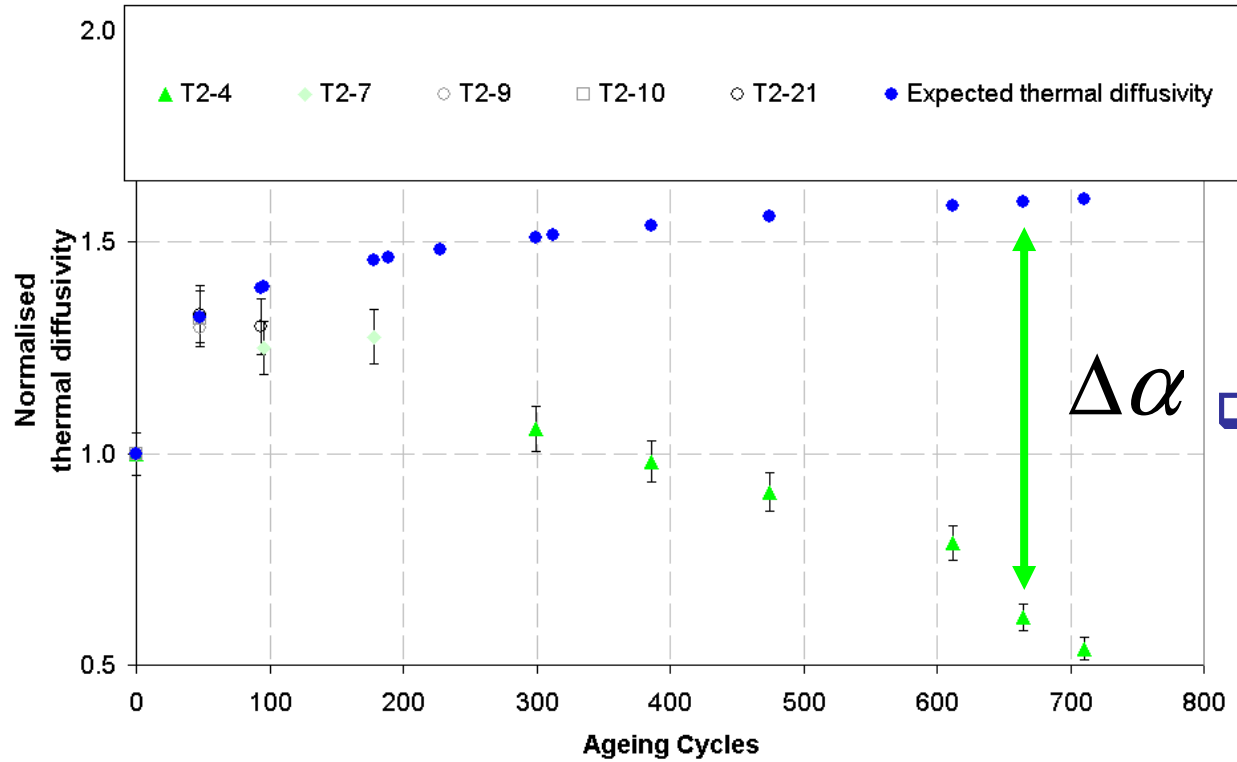


Crack growth ND detection on thermally cycled samples – Thick TBC



How to correlate the thermal diffusivity with the damage at the interface?





$\Delta\alpha$

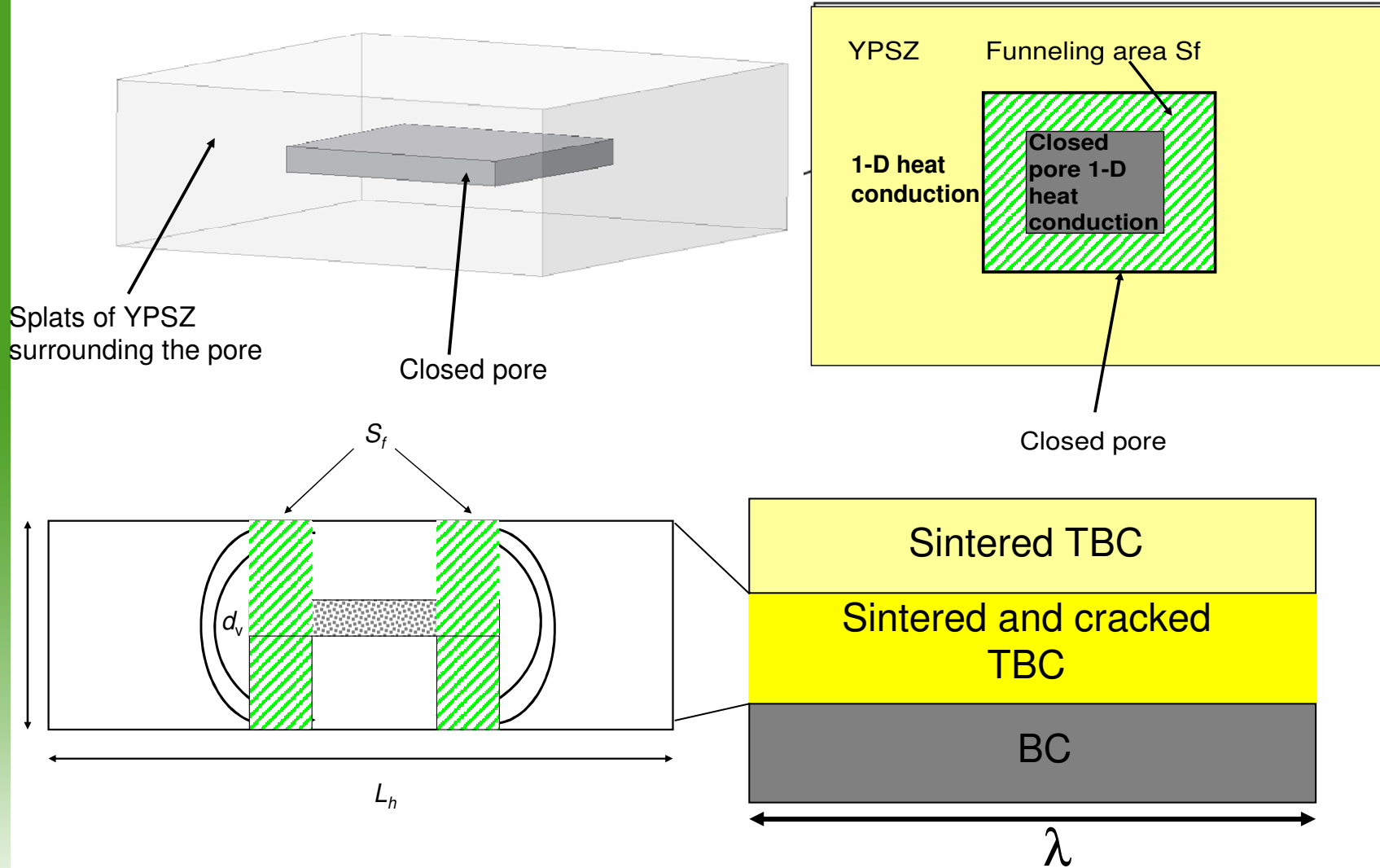


% of cracked interface



Modeling of the effect of cracks on TBC thermal diffusivity

Estimation of cracked interface by 2D- Inversion model

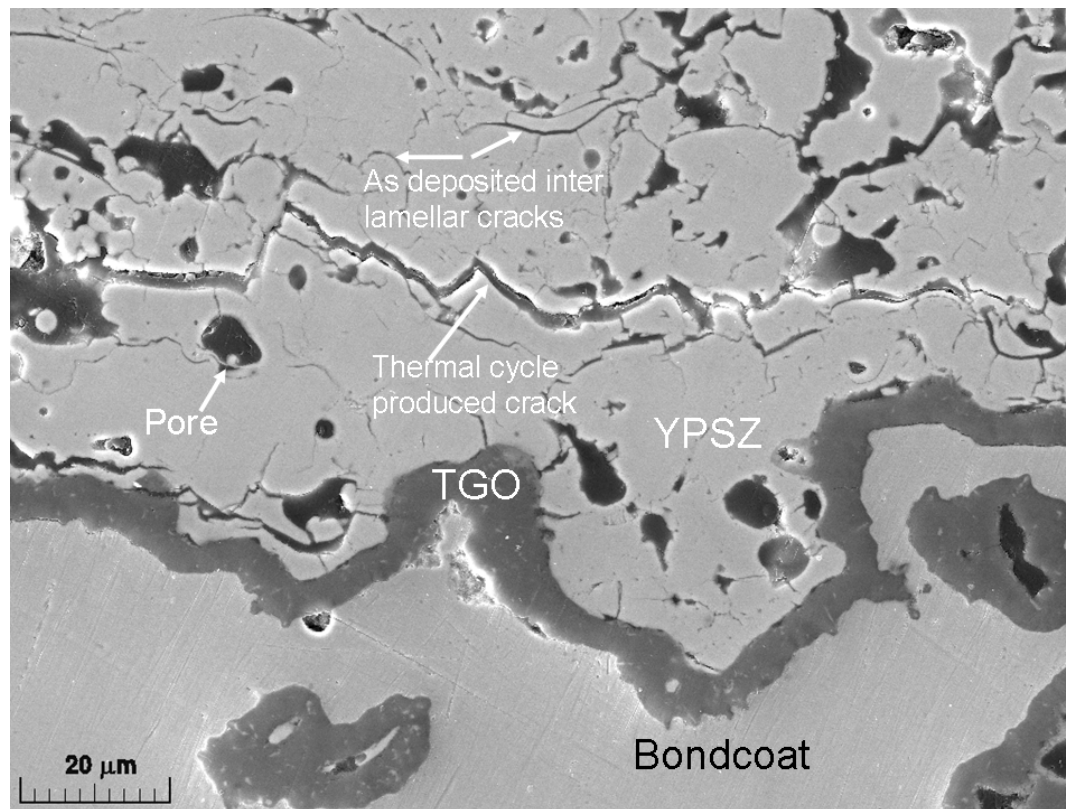


**How the cracked fraction of interface
has been estimated from IA?**

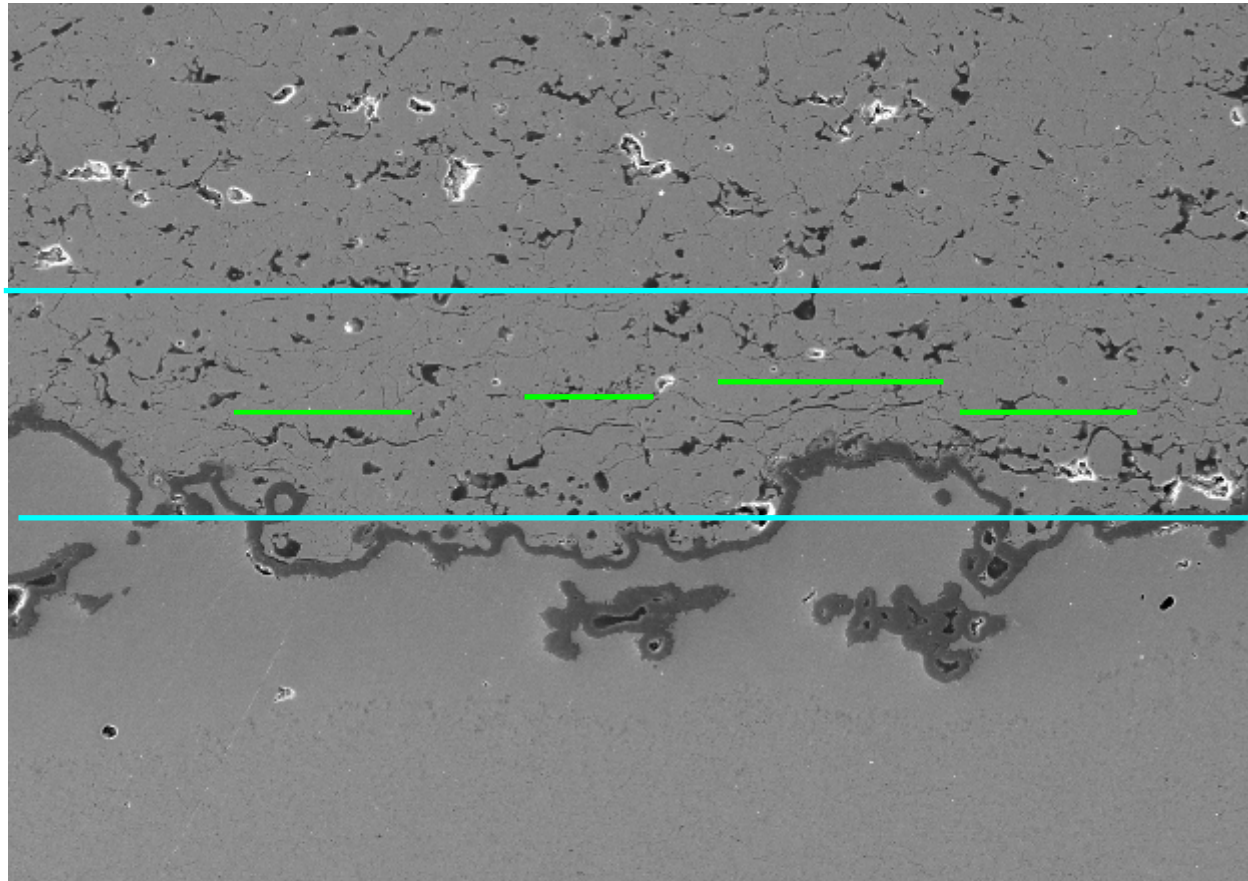
Estimation of cracked fraction of interface from IA



Cracks caused by thermal cycling have been supposed to be thicker and sharper than those originated during the spray process

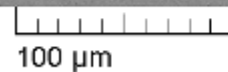


Estimation of cracked fraction of interface from IA



SEM HV: 20.00 kV
View field: 577.8 μm
Date(m/d/y): 02/18/10

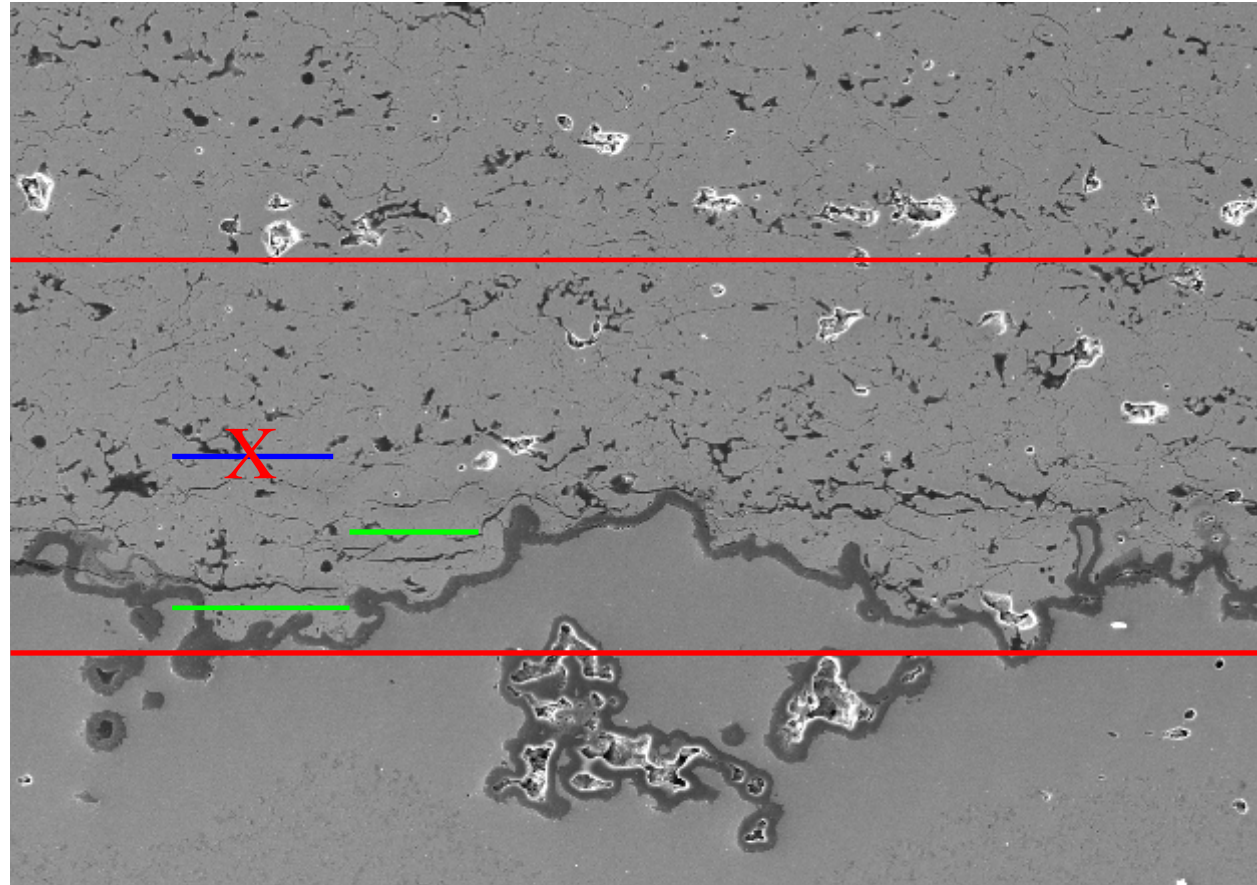
SEM MAG: 500 x
Det: SE



MIRAX TESCAN

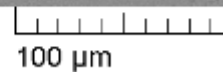
Performance in nanospace

Estimation of cracked fraction of interface from IA



SEM HV: 20.00 kV
View field: 577.8 μm
Date(m/d/y): 02/18/10

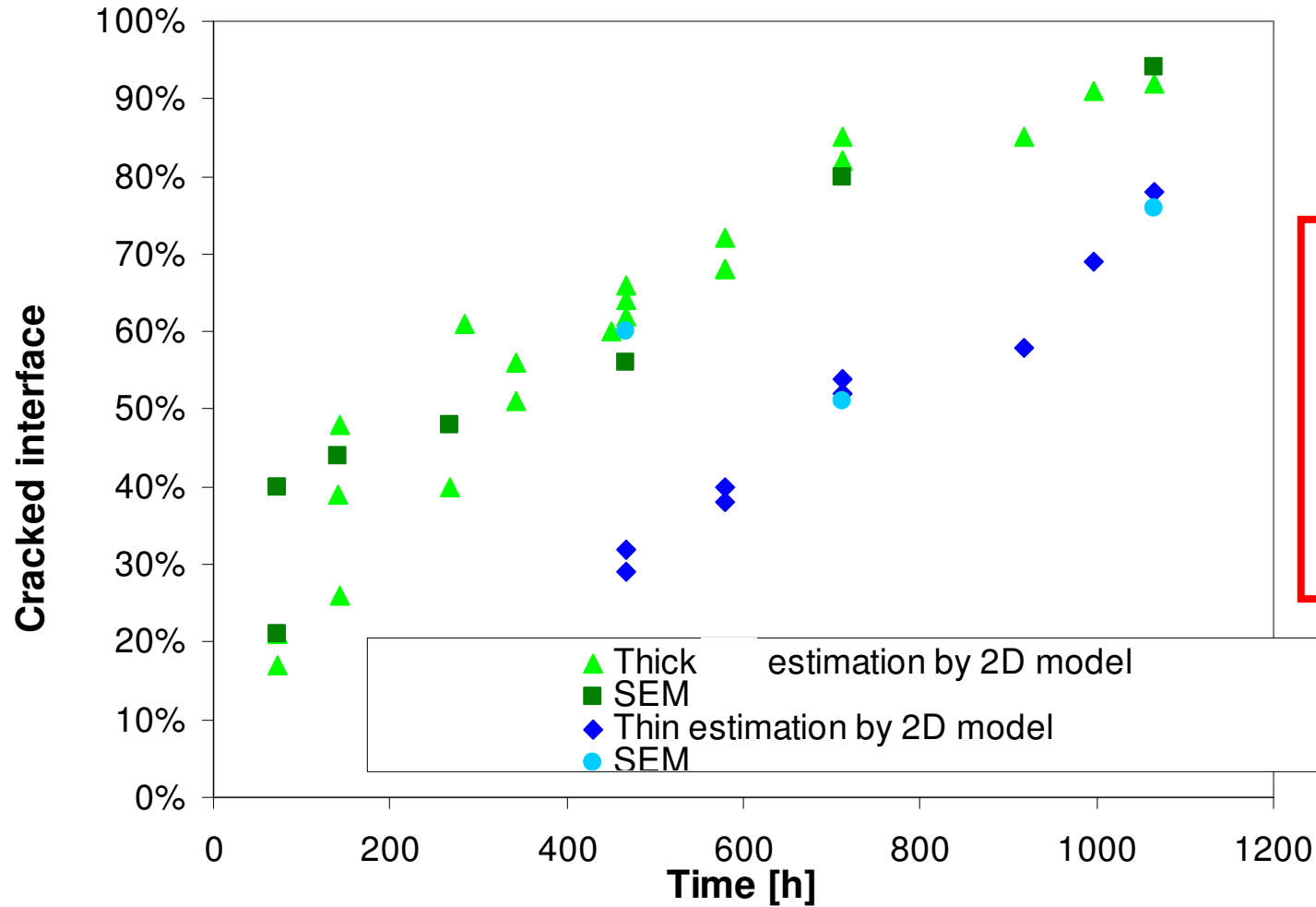
SEM MAG: 500 x
Det: SE



MIRA\\ TESCAN

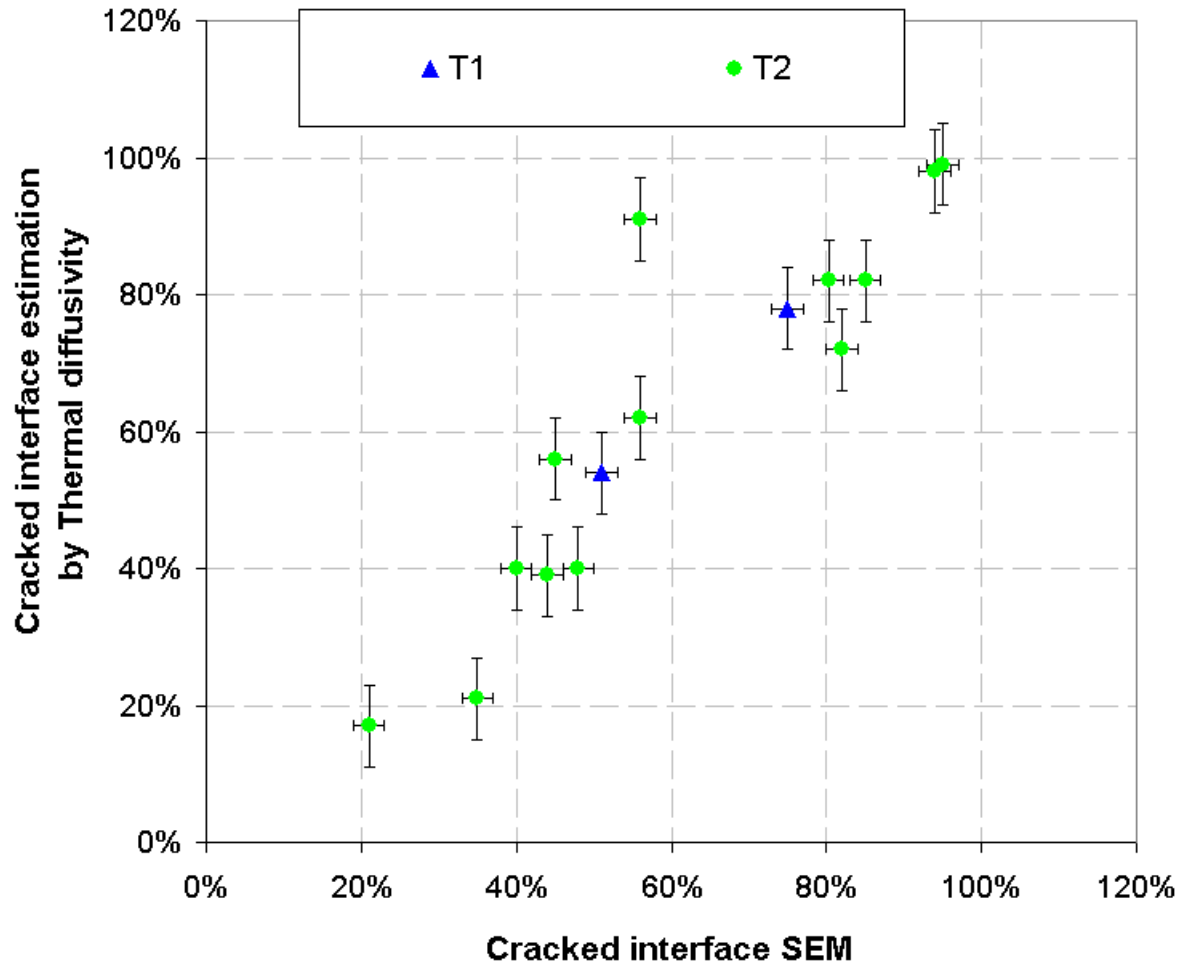
Performance in nanospace

Estimation of cracked interface by 2D- Inversion model



A good agreement between estimations and SEM is observed

Estimation of cracked interface by 2D- Inversion model



Estimation of cracked interface by 2D- Inversion model



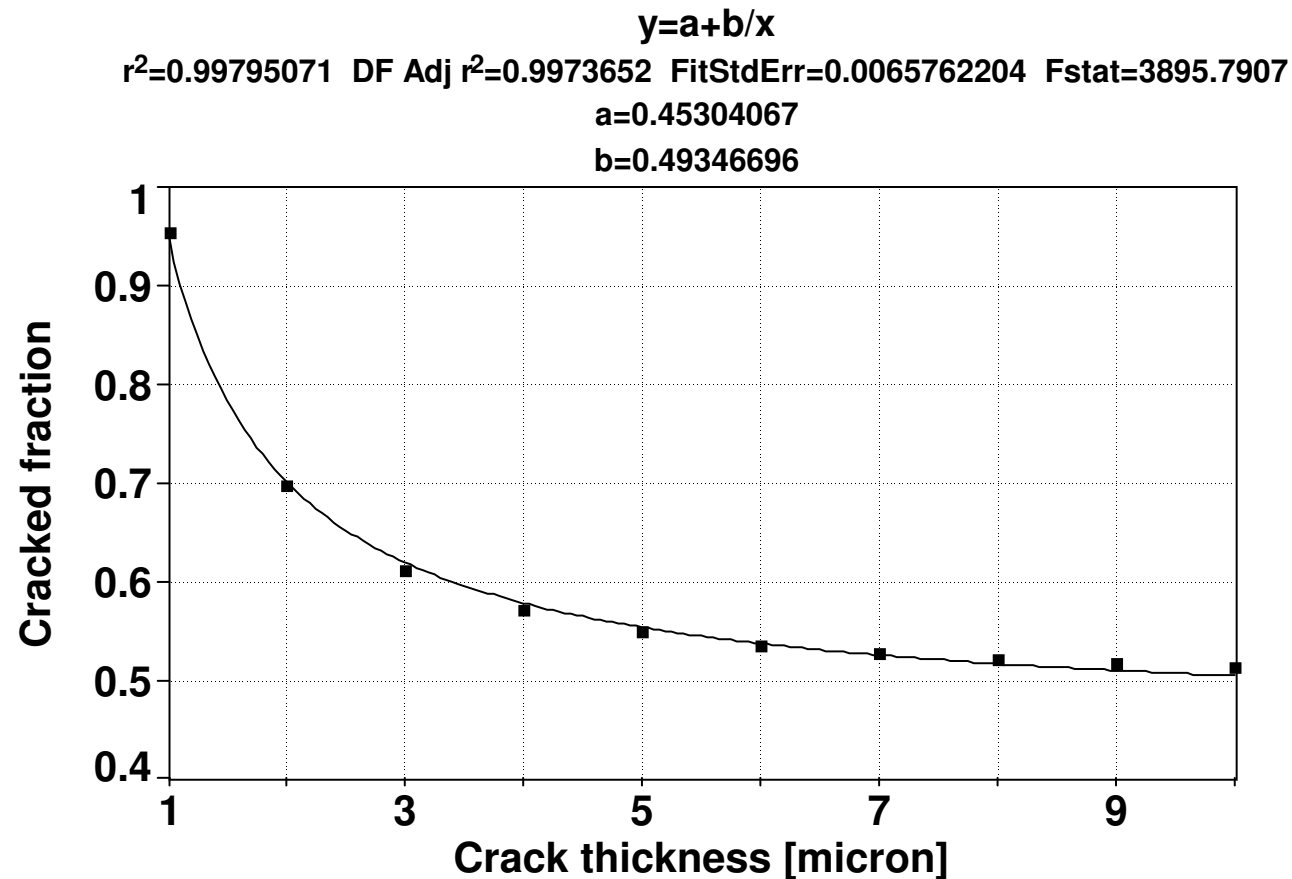
- The crack thickness has been fixed equal to 3 μm for each sample at each ageing time. When necessary, to zeroing the inversion function, the thickness has been increased up to values of 6 - 11 μm for ageing times close to the end of TBC life.
- For data referring to roughly 50% of TBC life, a crack thickness value intermediate between 3 μm and the maximum value was adopted.

How to avoid the “tuning” effect of the crack thickness in the inversion model?

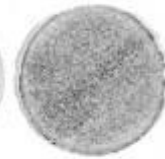
A figure of merit



Consider the trend of cracked fraction vs. crack thickness and to look for a simple fitting function whose free parameters can be related to the damage at the interface

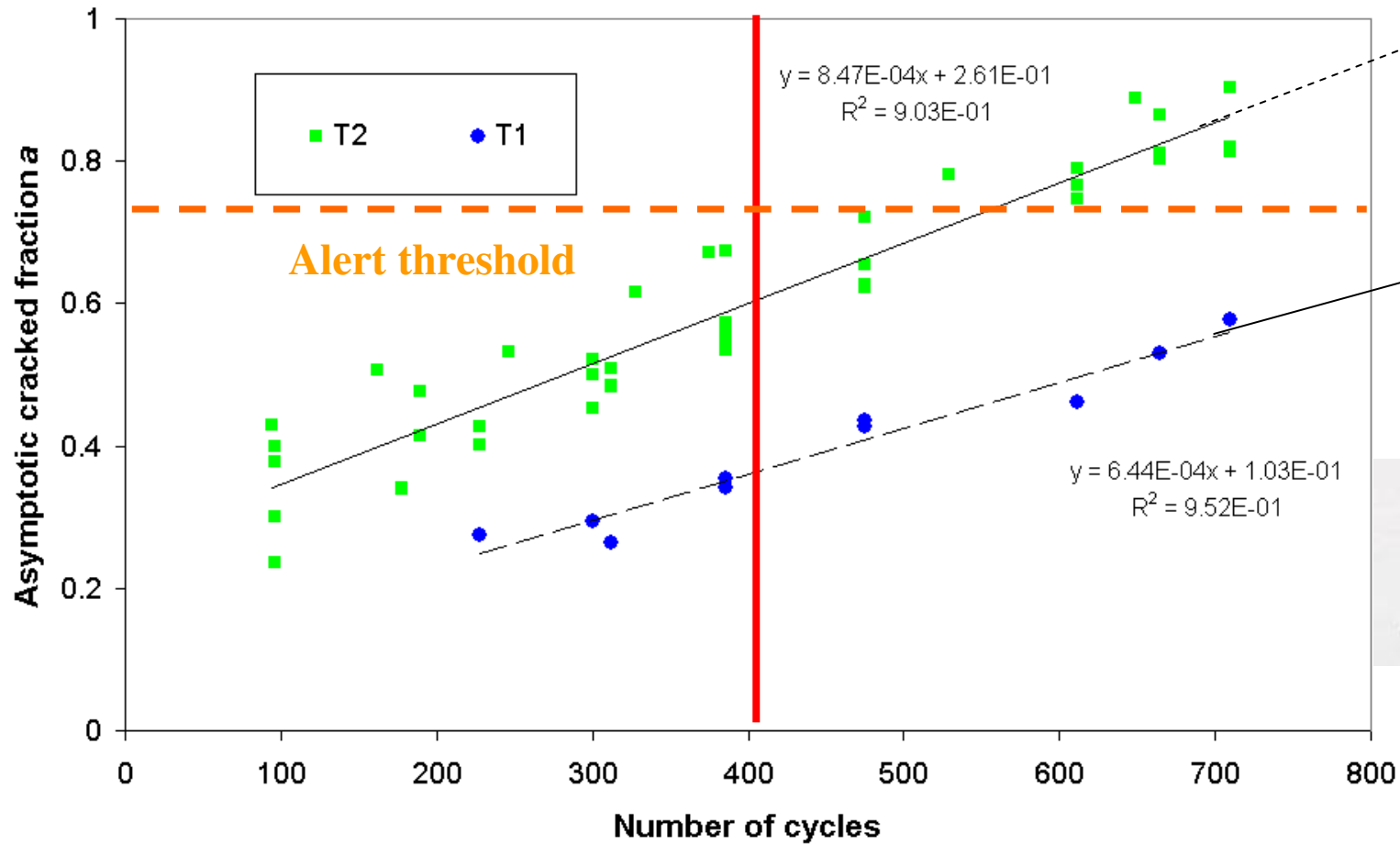


A figure of merit



$8.0 \pm 0.9 \cdot 10^{-4}$

872 cycles



1392 cycles



Conclusions

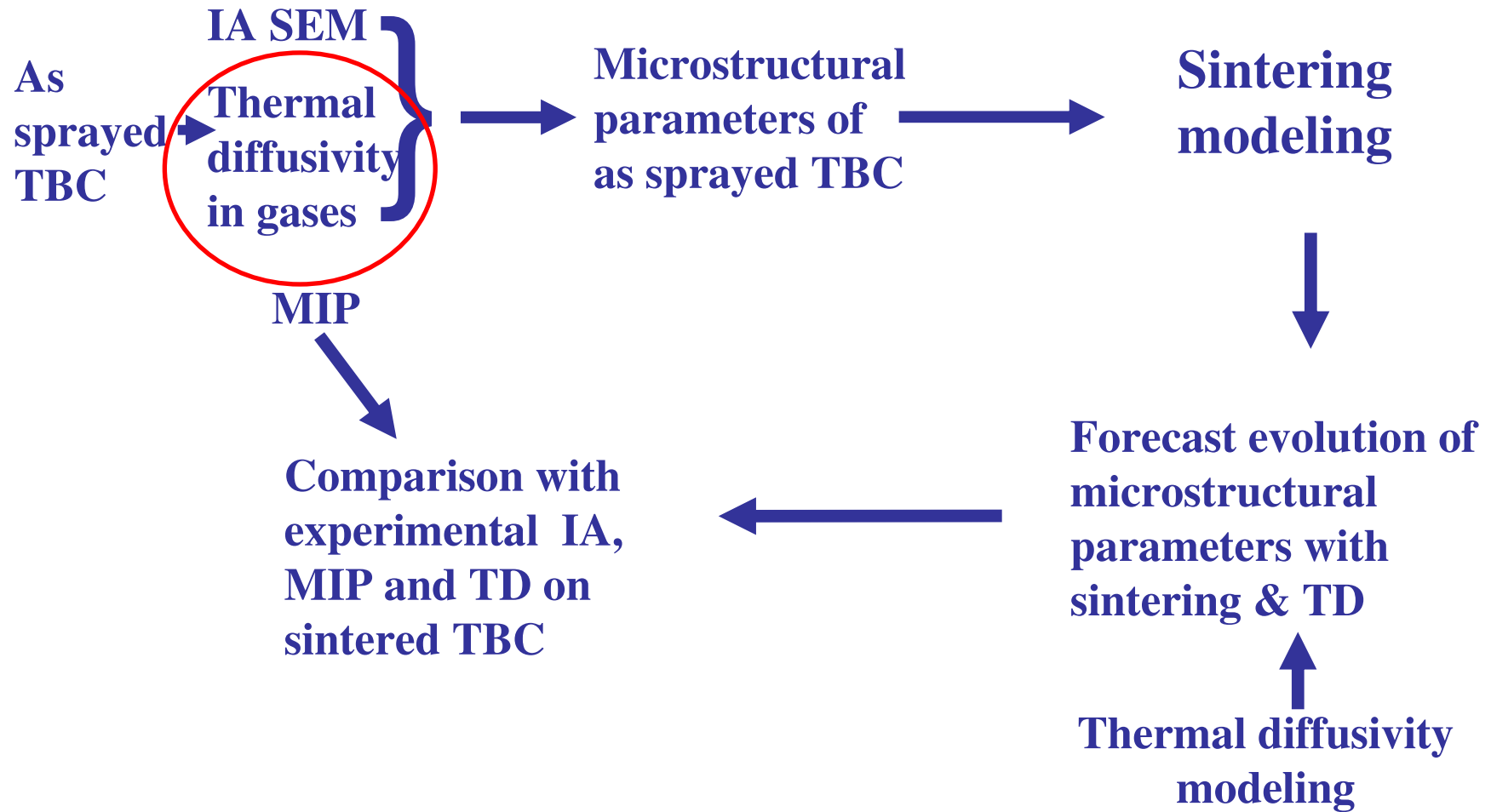


- A 2D model has been used for better estimating the cracked fraction of the interface between APS TBC and BC, especially at early stage when there are several short cracks.
- A figure of merit incorporating both cracked fraction and crack thickness was also proposed for ranking the damage and obtaining indications on the failure time

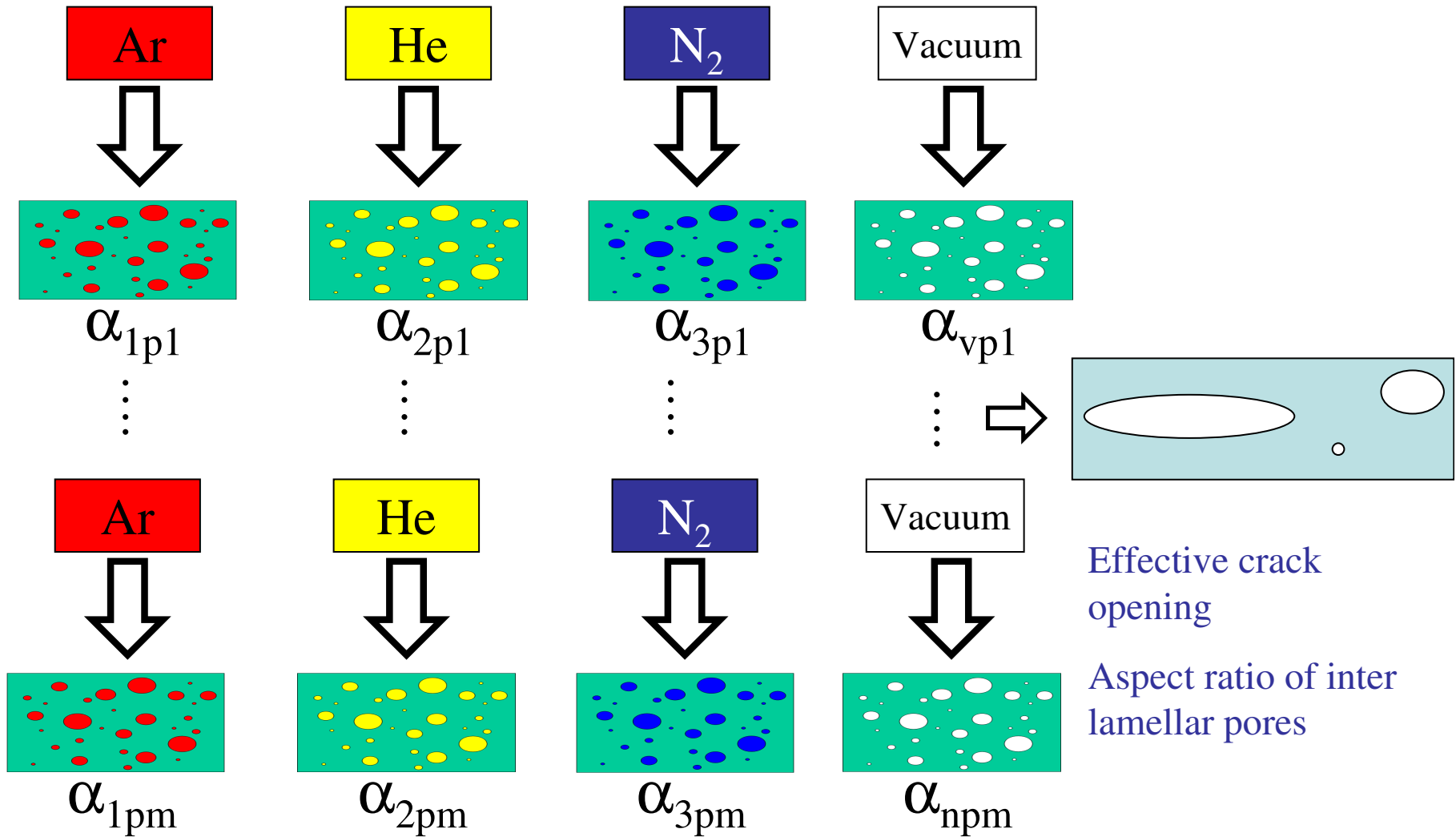
*Microstructural characterisation of APS TBC by
thermal diffusivity evaluation and sintering
forecasts*

F. Cernuschi, I.G. Golosnoy, P. Bison, A. Moscatelli, R. Vassen, H-P Bossmann and S. Capelli, “Microstructural Characterization of Porous Thermal Barrier Coatings by IR gas porosimetry and sintering forecasts” submitted to Acta Materialia

The idea

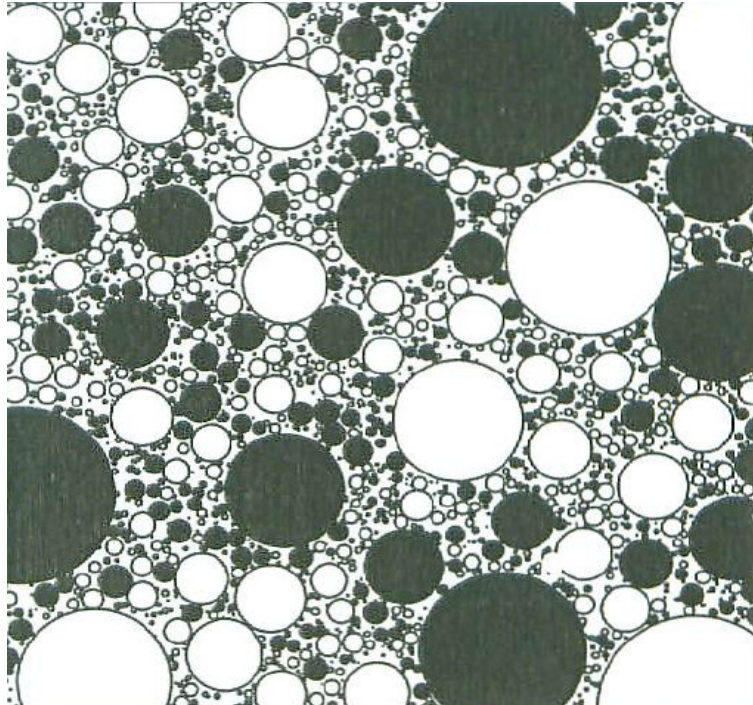


Thermal diffusivity in different gases

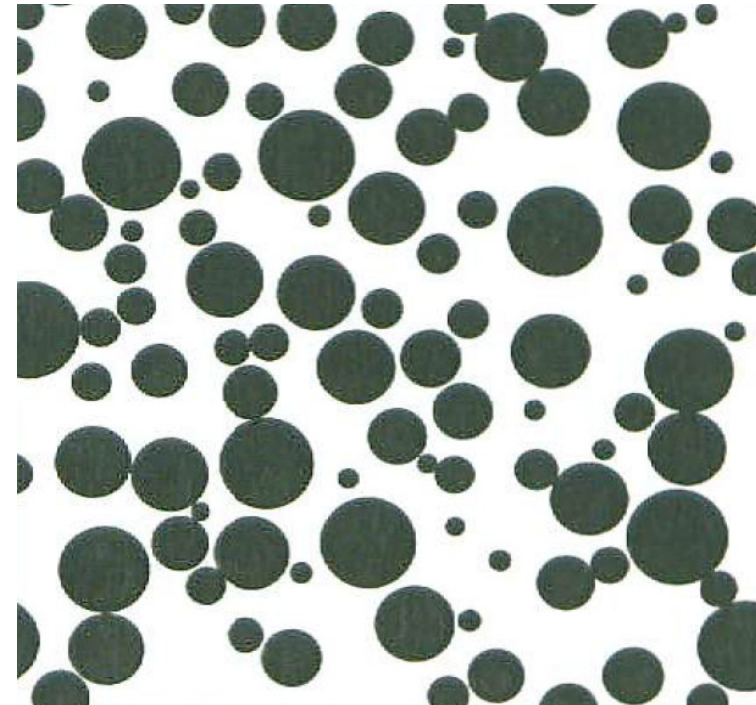


Modelling of porous solids

Symmetric models



Asymmetric models



TBC



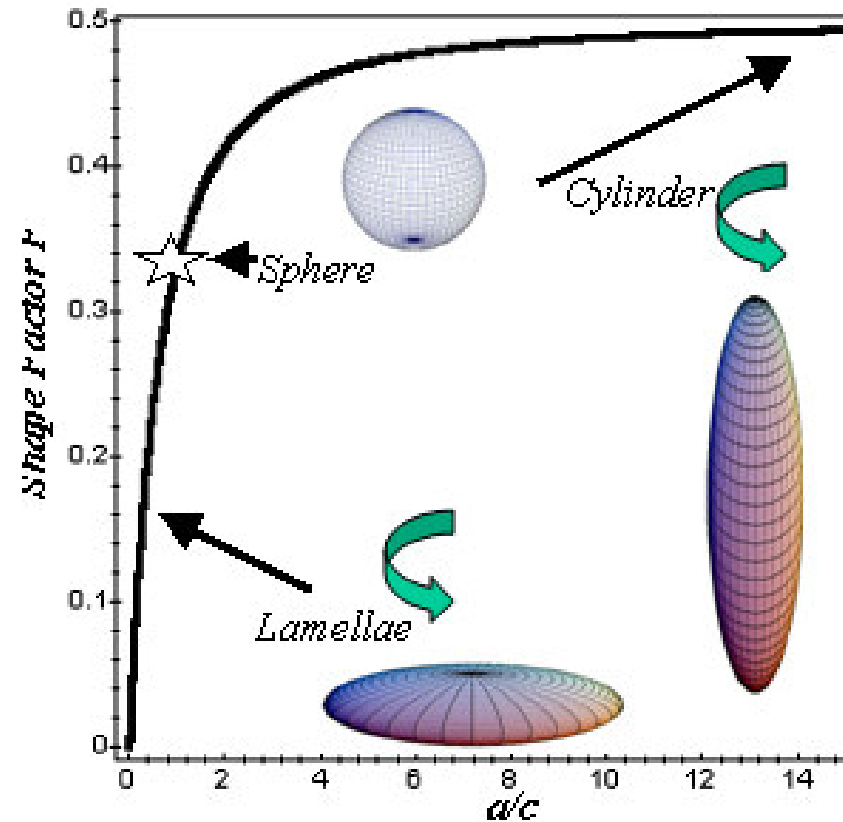
Asymmetric model



Modelling of porous solids

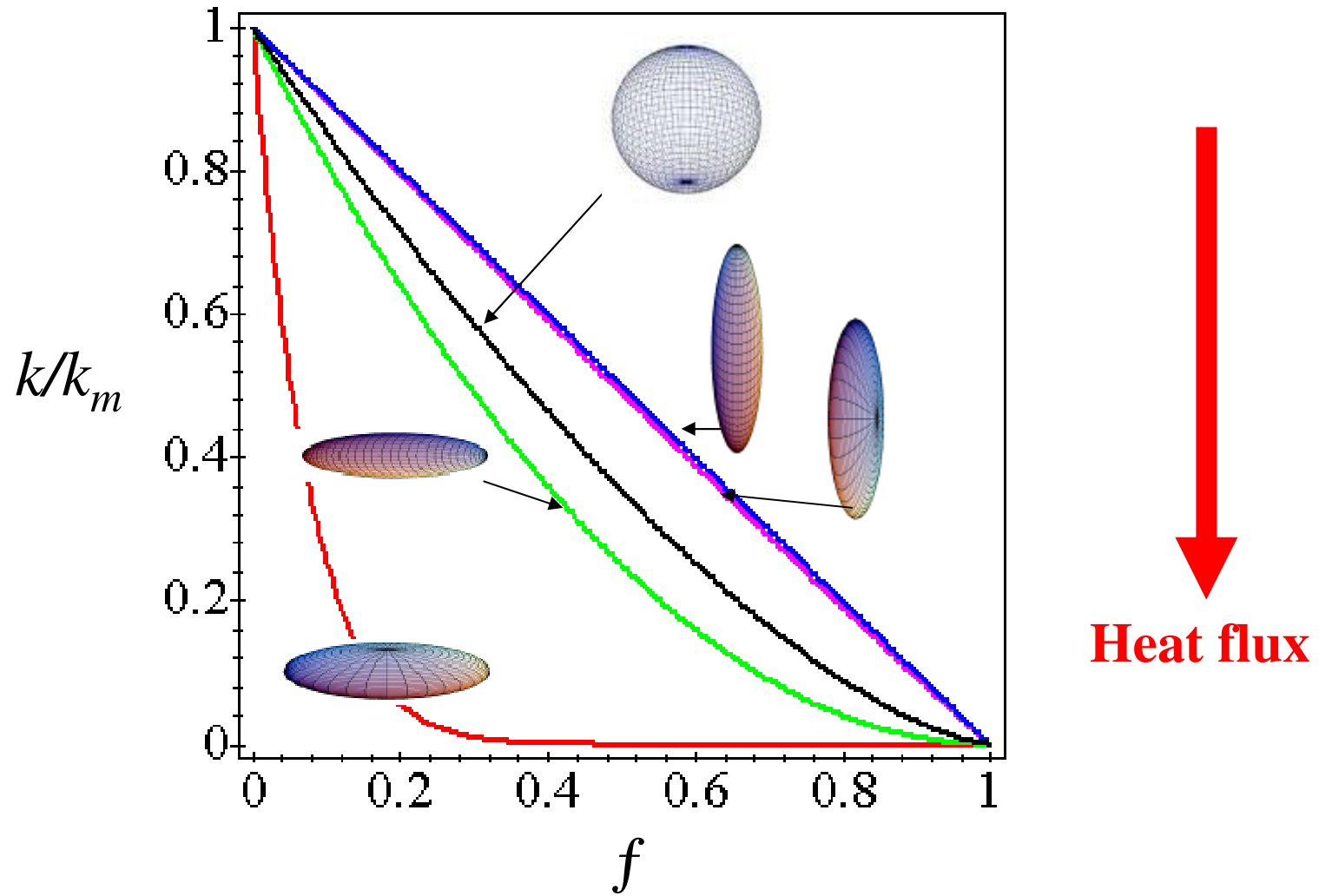
$$\frac{k}{k_m} = (1 - f)^X$$

$$X = \frac{1 - \cos^2 \alpha}{1 - F} + \frac{\cos^2 \alpha}{2F}$$



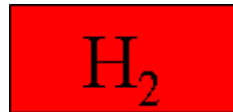
D.A.M Bruggeman, Annalen der Physik n°5 [24] 1935

Modelling of porous solids



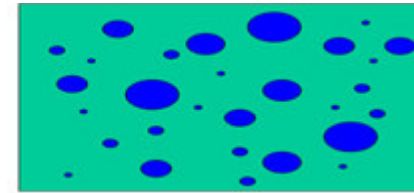
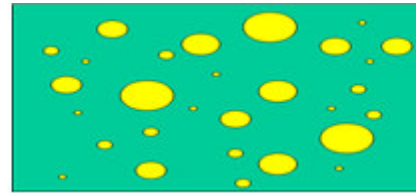
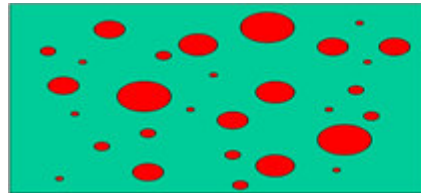
Modelling of porous solids - thermal conductivity of gases within pores

k_{bulk}

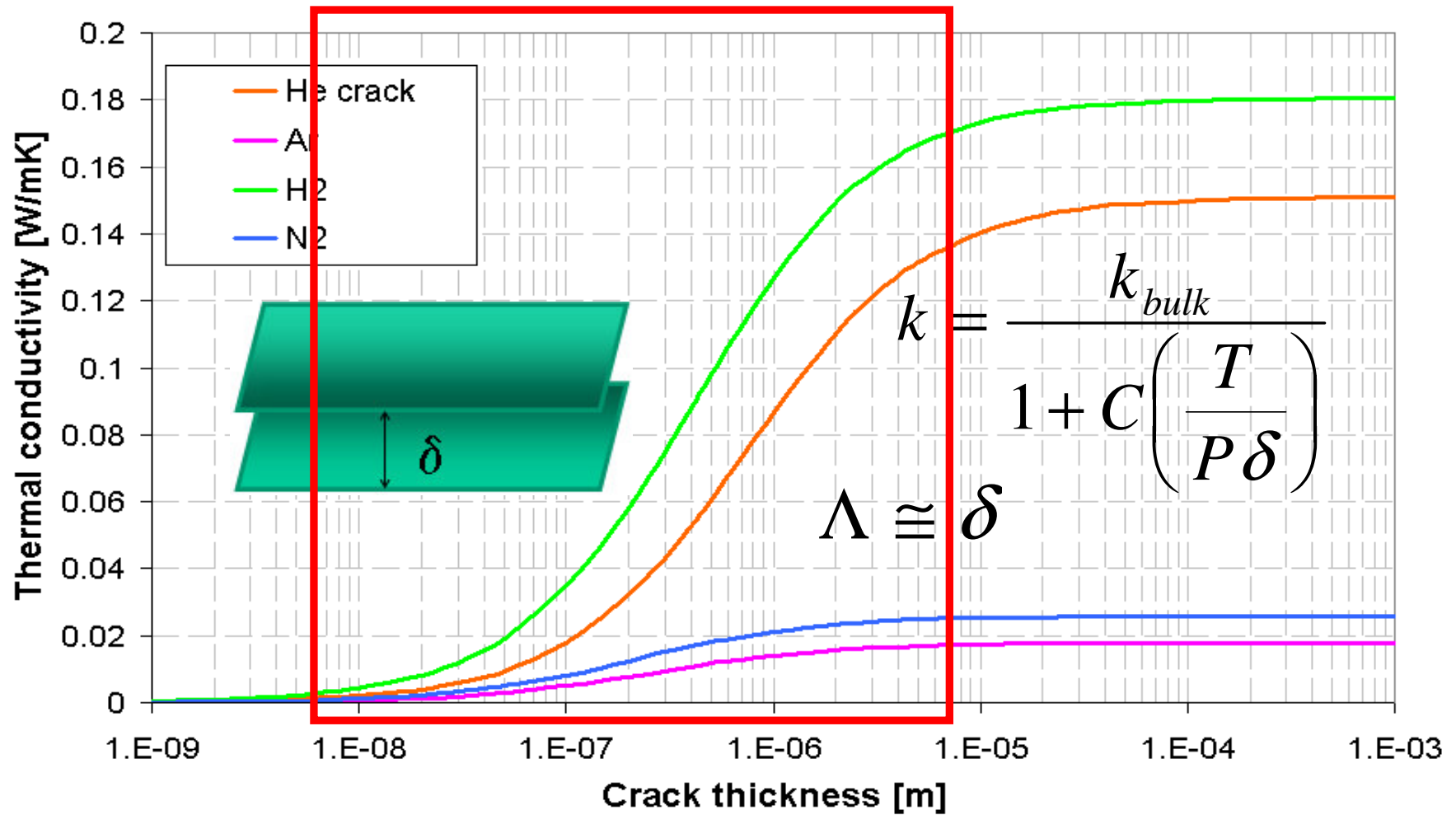


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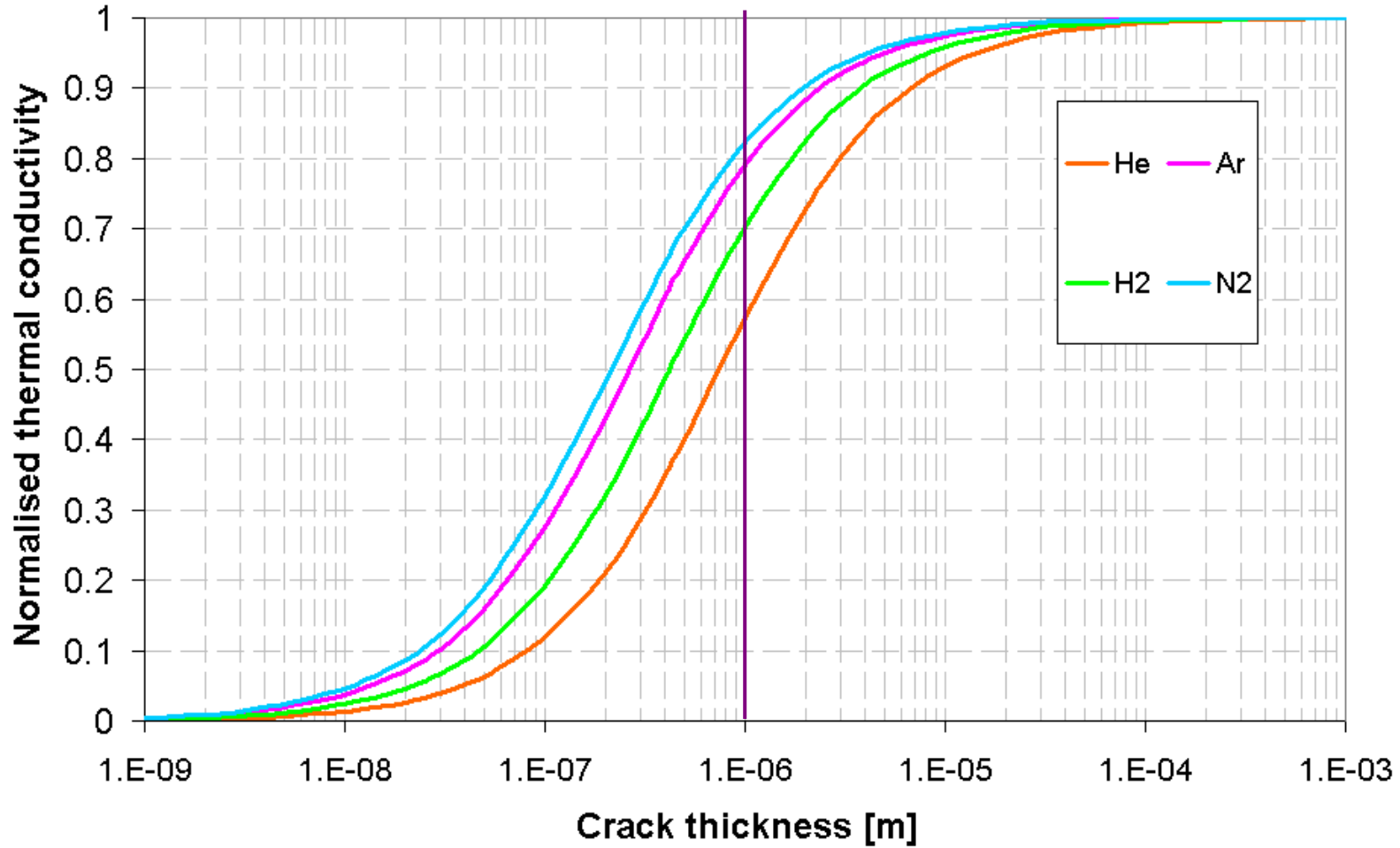
k_{pores}



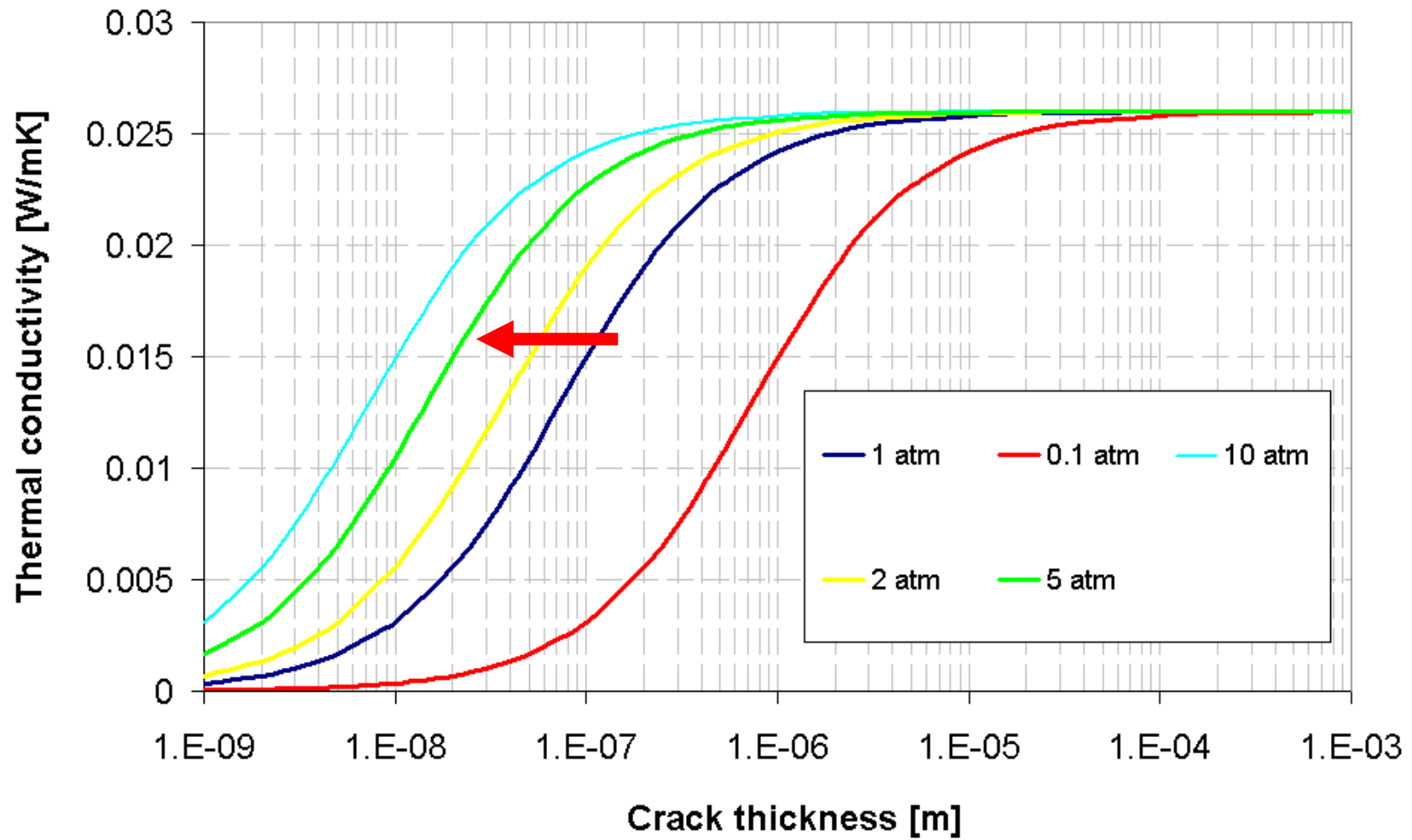
Modelling of porous solids - thermal conductivity of gases within pores



Modelling of porous solids - thermal conductivity of gases within pores

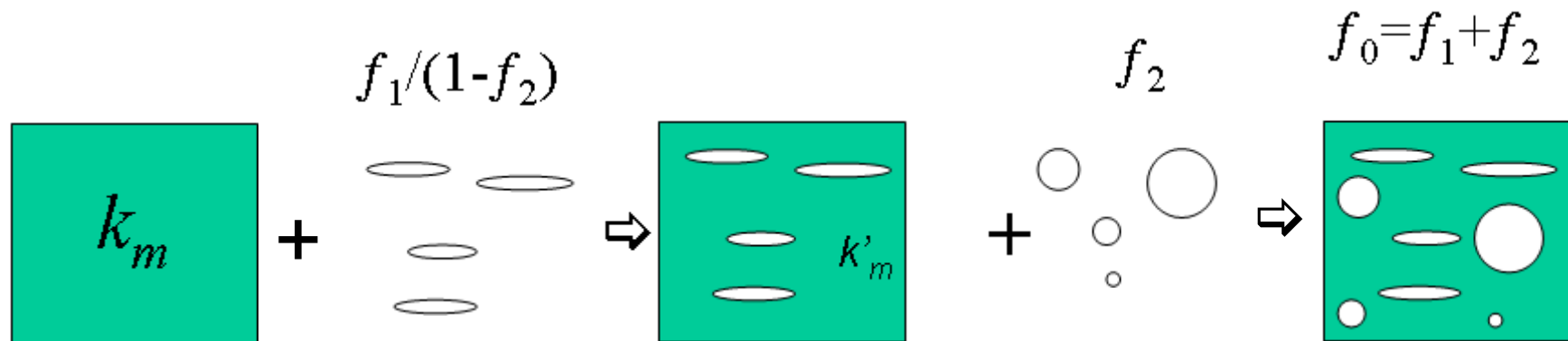


Modelling of porous solids - thermal conductivity of gases within pores



Modelling of porous solids

How is it possible to simultaneously model several types of pores?



$$k = \frac{1}{2} \left\{ \Phi \left(\Psi \left(k_m, \frac{f_1}{(1-f_2)} \right), f_2 \right) + \Psi \left(\Phi \left(k_m, \frac{f_2}{(1-f_1)} \right), f_1 \right) \right\}$$

The samples - YPSZ APS TBC

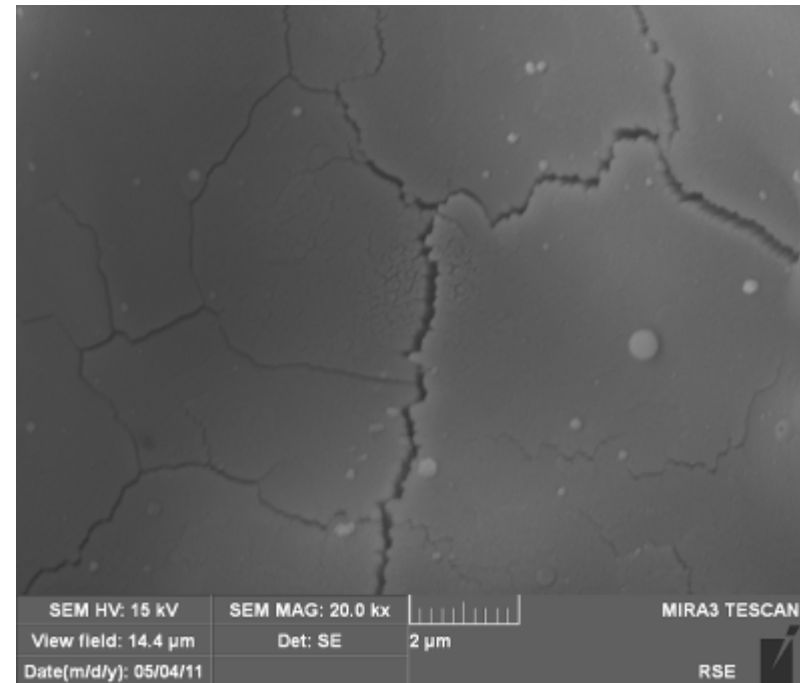
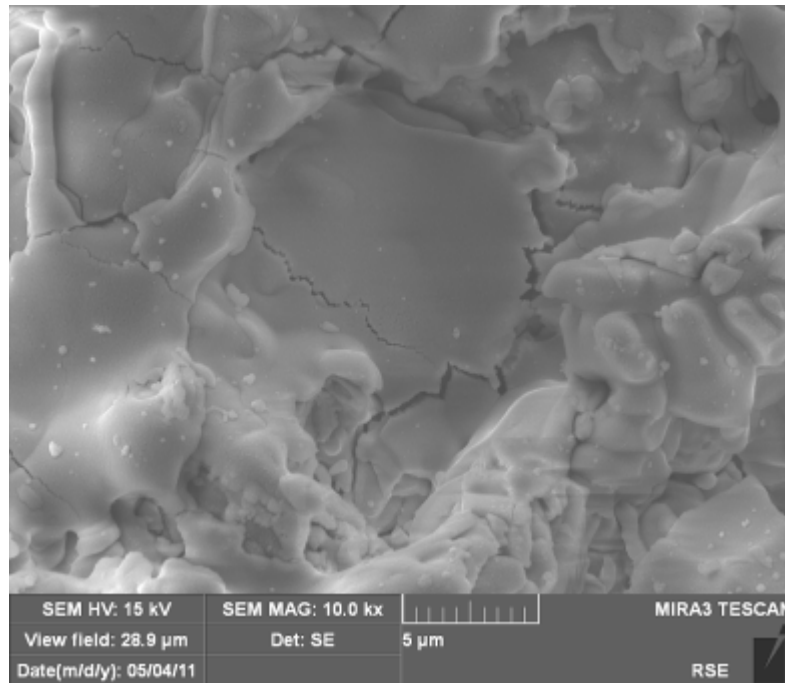


Sample Code	Thickness [μm]	Shape	Size [mm]	Heat treated	Characterised by
WD1	427 \pm 16	disk	9.81	Y	Thermal diffusivity and SEM
WD2	416 \pm 15	disk	9.81	Y	Thermal diffusivity and SEM
WD3	400 \pm 21	disk	9.81	Y	Thermal diffusivity and SEM
WD4	414 \pm 13	Square	20	N	SEM
WD6	n.a.	Rectangular	10 X40	N	Hg intrusion Porosimetry
WD7	n.a.	Rectangular	10 X40	N	Hg intrusion Porosimetry
WD8	n.a.	Rectangular	10 X40	Y	Hg intrusion Porosimetry
WD9	n.a.	Rectangular	10 X40	Y	Hg intrusion Porosimetry

As sprayed

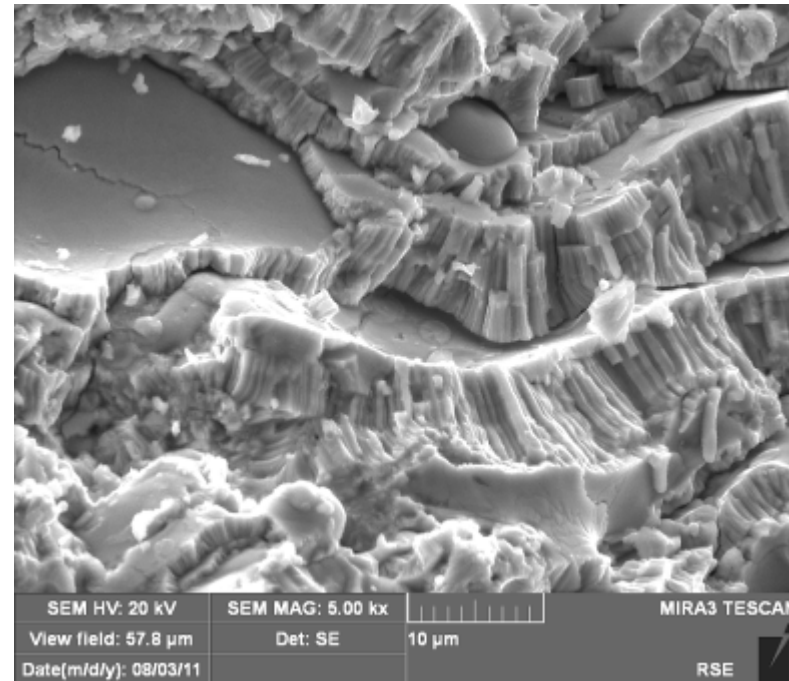
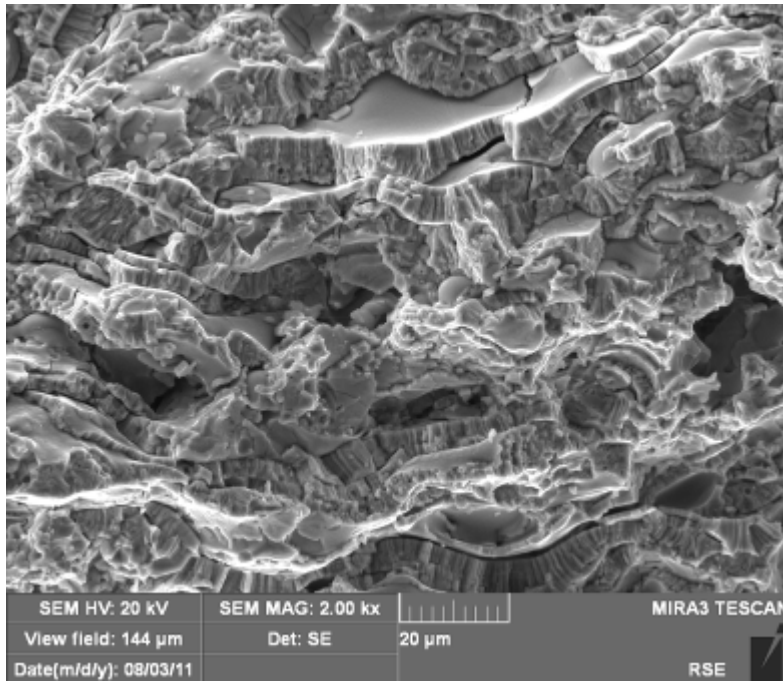
Heat treated 200 h @ 1250°C

The samples - YPSZ APS TBC



Vertical intra-lamellar cracks

The samples - YPSZ APS TBC



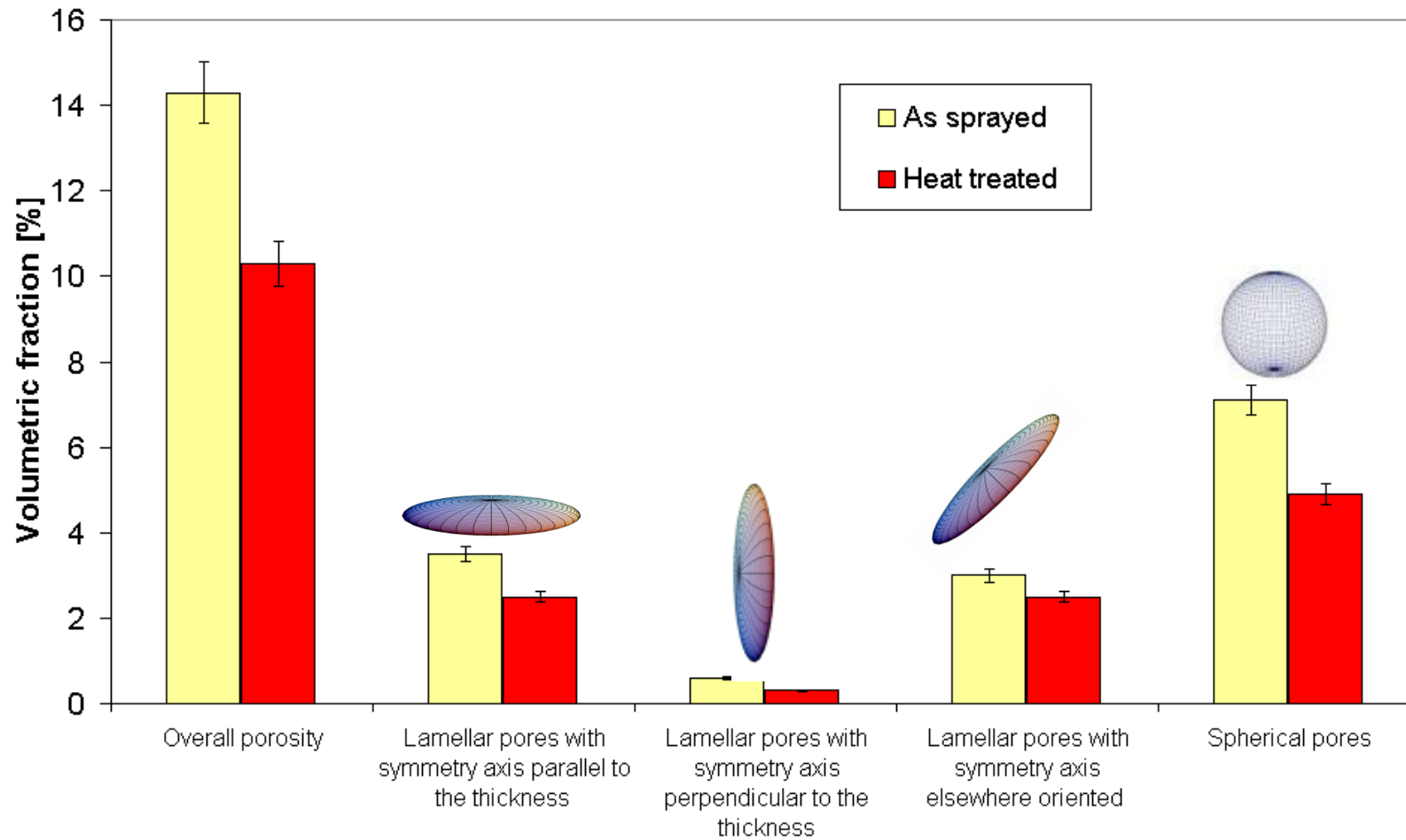
Horizontal inter-lamellar pores

Vertical intra-lamellar cracks

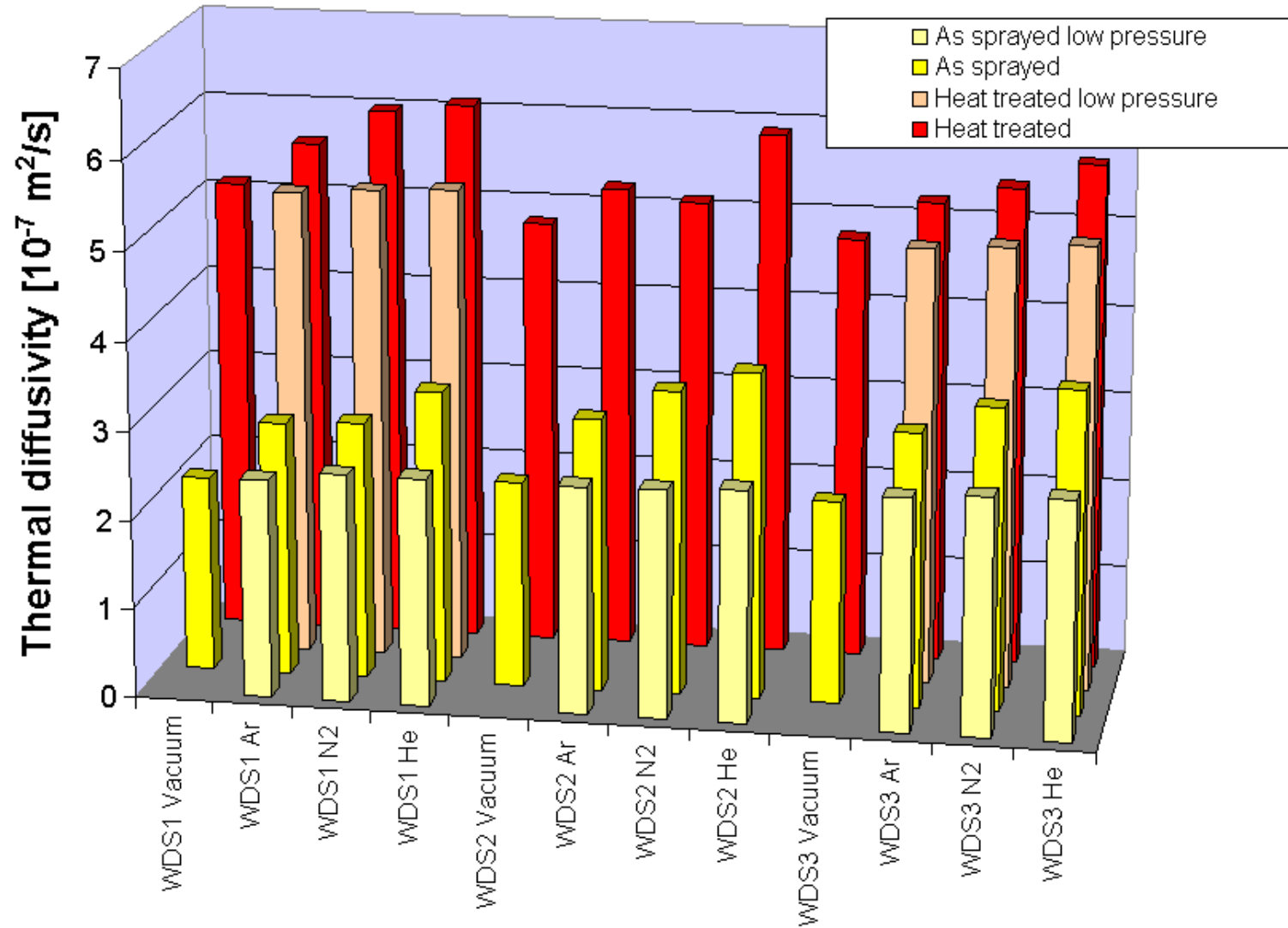
Globular pores

IA characterisation

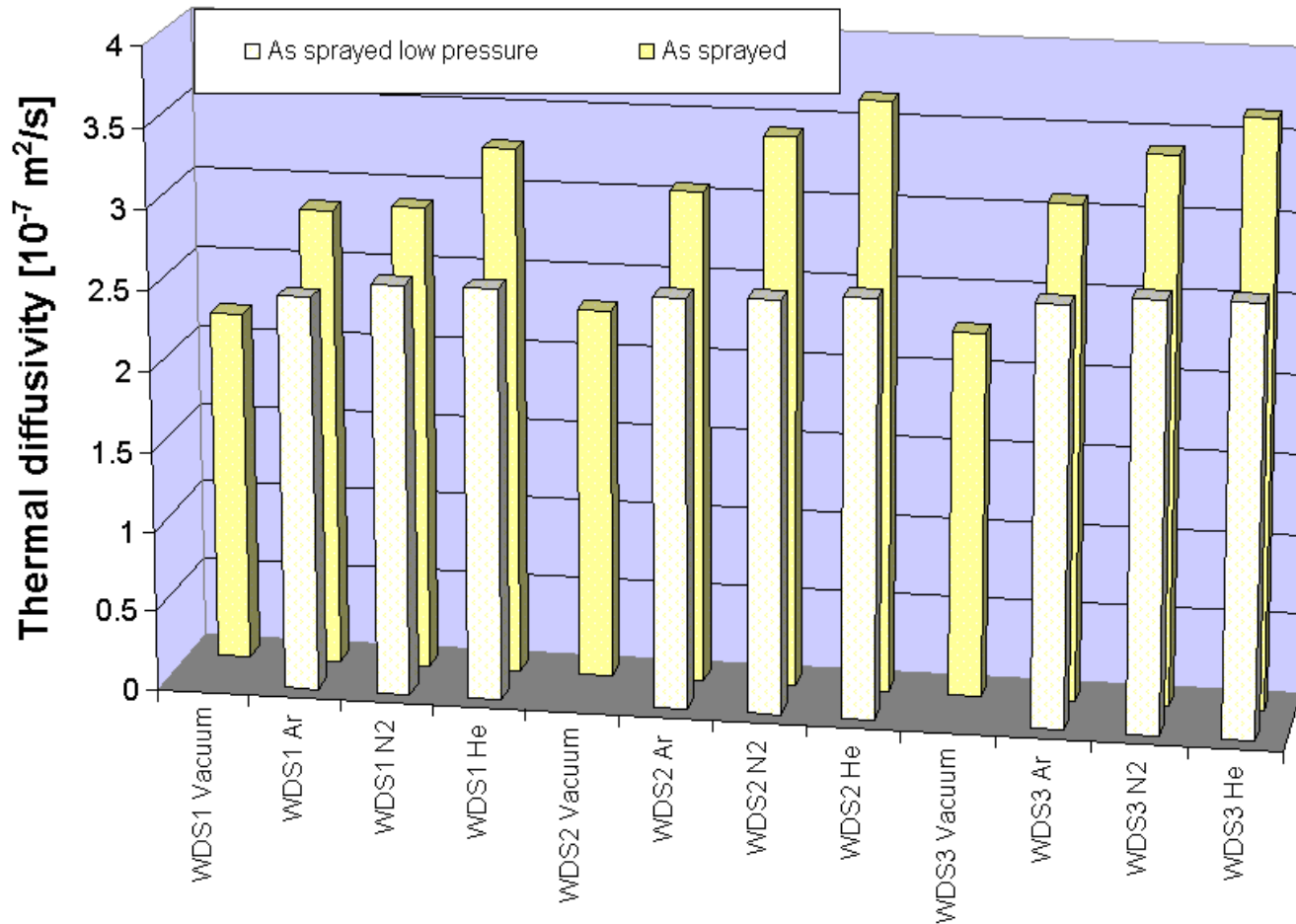
Elongation ($e > 3$) and orientation criteria



Experimental thermal diffusivity data



Experimental thermal diffusivity data



Input of the inversion code

As lamellar porosity parallel oriented to the heat flux **plays the major role in reducing** the thermal diffusivity, starting from the IA volumetric fractions, in the inversion code, porosity has been divided into two classes: parallel inter-lamellar pores and spheres.

Inter-lamellar parallel = true inter-lamellar parallel + 1/3 elsewhere oriented lamellar

4.1% 1%

Spheres = True spheres + 2/3 elsewhere oriented lamellar + vertical intra-lamellar cracks.

6.15%

2%

0.55%

5.1% + 8.7% = 13.8%



7 equations for each sample (vacuum , three gases, two pressures)

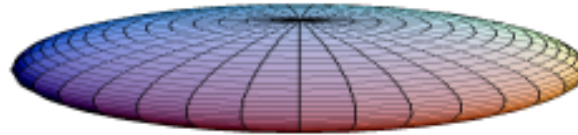


Average crack thickness and aspect ratio

Output of the inversion code

WDS1 as Sprayed

$$a/c=1/38$$



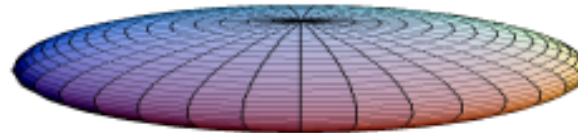
Specific surface area: $1.82 \cdot 10^6 \text{ m}^2/\text{m}^3$

Porosity 13.8
 $0.088 \pm 0.005 \text{ }\mu\text{m}$

Porosity 13.1

WDS2 as Sprayed

$$a/c=1/37$$

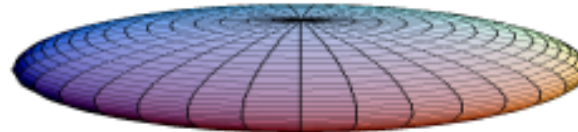


Specific surface area: $1.0 \cdot 10^6 \text{ m}^2/\text{m}^3$

$0.163 \pm 0.008 \text{ }\mu\text{m}$

WDS3 as Sprayed

$$a/c=1/37$$

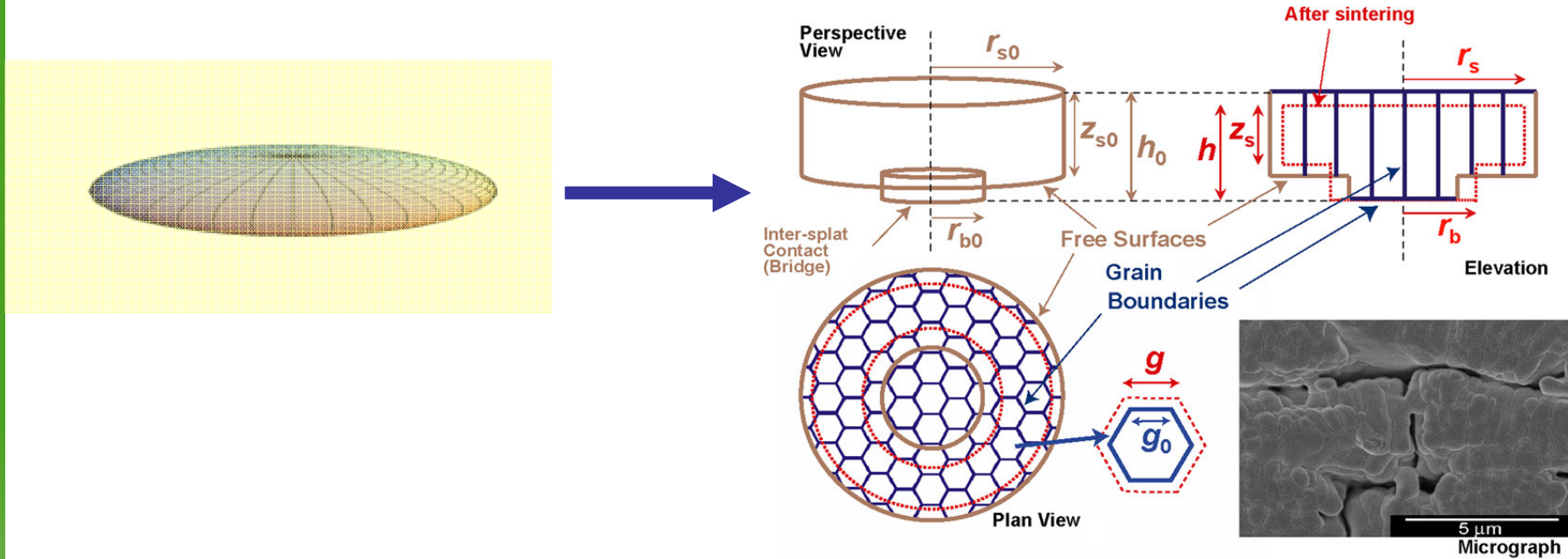


Specific surface area: $1.0 \cdot 10^6 \text{ m}^2/\text{m}^3$

$0.163 \pm 0.008 \text{ }\mu\text{m}$

Porosity 14.4

Input for the sintering code



Bruggeman Model



Cipitria Model

Translation

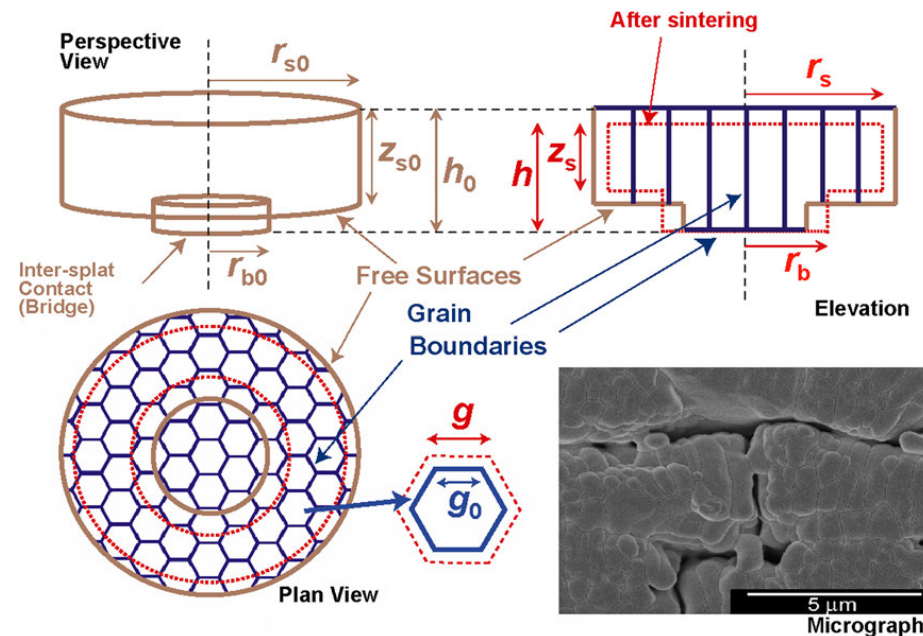
The Cipitria's sintering code

Surface and grain boundary diffusion, together with grain boundary migration

Grain boundary diffusion, leading to through-thickness shrinkage (reduction in r_s , h),

Surface diffusion contributes to pore spheroidization, i.e. reduces z_s , causing the half-height of the pore ($h - z_s$) to increase, and increases r_b .

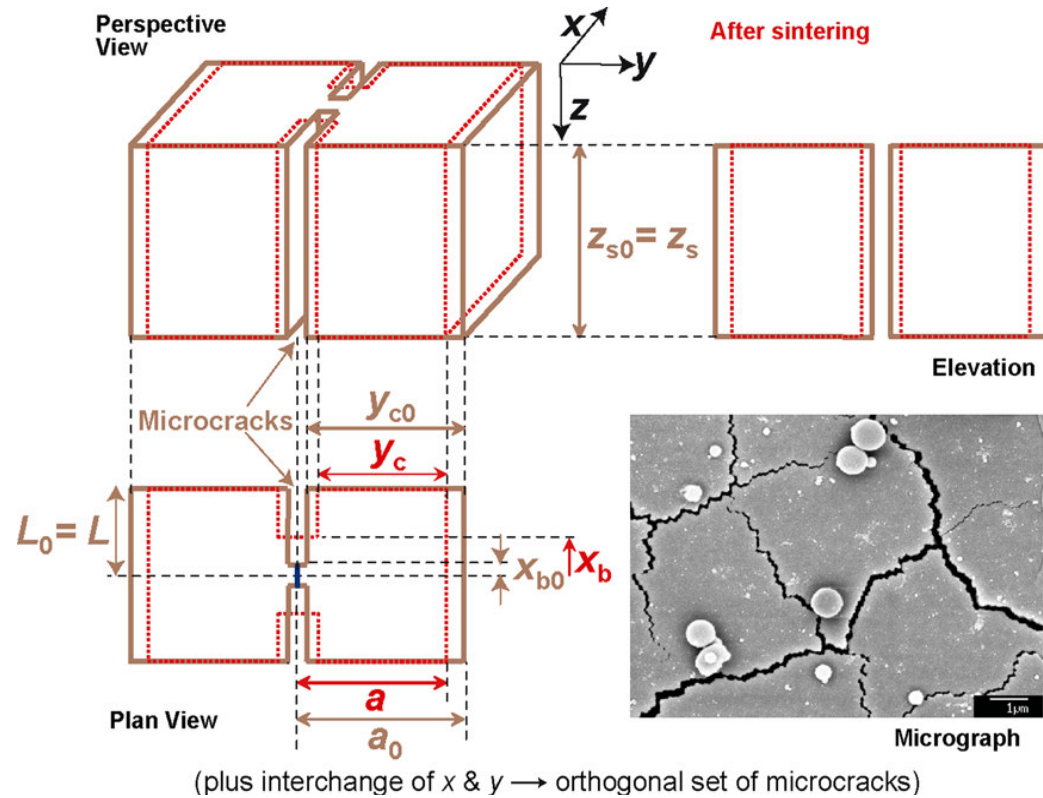
Grain boundary migration causes an increase in lateral (in-plane) grain size, g , and hence a reduction in N_s



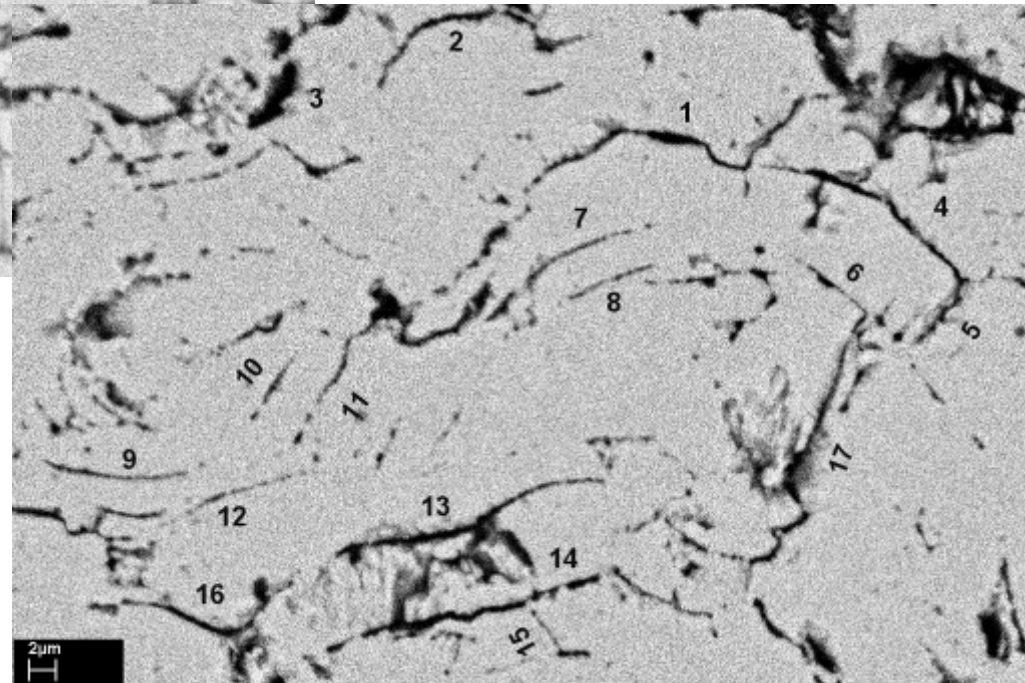
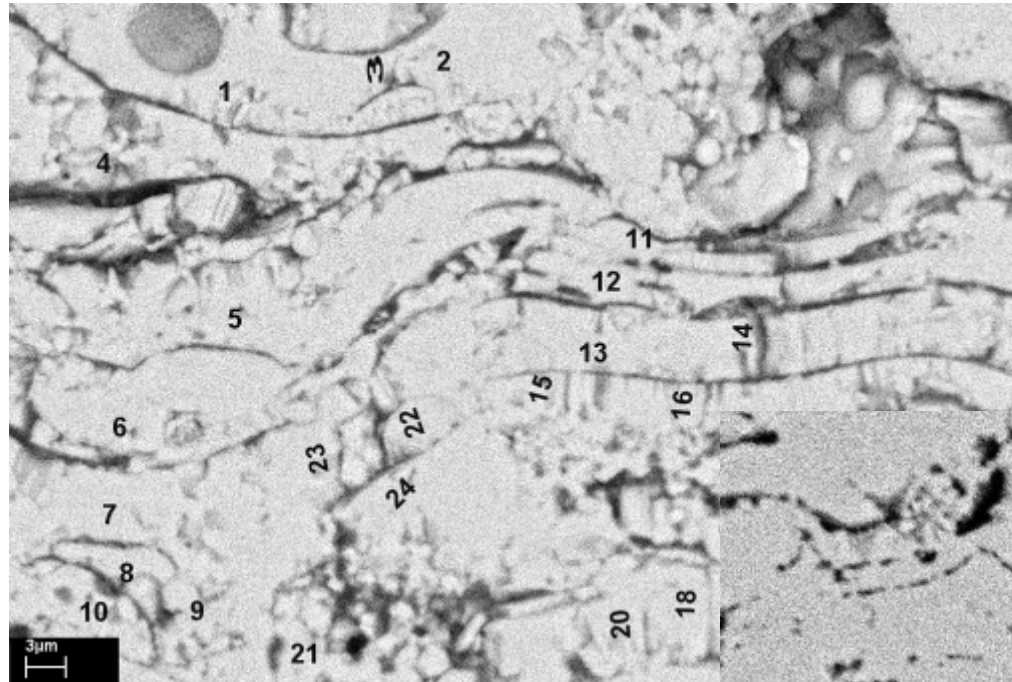
The Cipitria's sintering code

Grain boundary diffusion
(reduction in a and induced increase in x_b)

Surface diffusion
spheroidization of microcracks, i.e. reduces y_c , which promotes opening of the microcracks ($(a-y_c)$ increases), and increases x_b .



IA characterisation



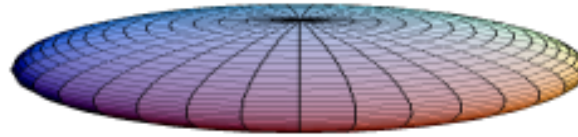
Output of the **inversion** and **sintering** codes

WDS1 heat treated

$$a/c=1/38$$

$$a/c=1/22$$

$$a/c=1/6$$



$$0.088 \pm 0.005 \mu\text{m}$$

$$0.37 \pm 0.05 \mu\text{m}$$

$$0.26 \pm 0.05 \mu\text{m}$$

$$\text{Specific surface area: } 1.82 \cdot 10^6 \text{ m}^2/\text{m}^3$$

$$\text{Specific surface area: } 1.76 \cdot 10^6 \text{ m}^2/\text{m}^3$$

$$\text{Specific surface area: } 0.39 \cdot 10^6 \text{ m}^2/\text{m}^3$$

$$\text{Specific surface area: } 0.58 \cdot 10^6 \text{ m}^2/\text{m}^3$$

Porosity 10.1

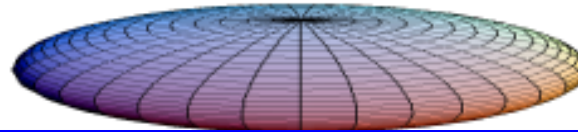
Porosity 8.8

WDS3 heat treated

$$a/c=1/37$$

$$a/c=1/25$$

$$a/c=1/14$$



$$0.163 \pm 0.008 \mu\text{m}$$

$$0.22 \pm 0.05 \mu\text{m}$$

$$0.30 \pm 0.05 \mu\text{m}$$

$$\text{Specific surface area: } 0.55 \cdot 10^6 \text{ m}^2/\text{m}^3$$

$$\text{Specific surface area: } 0.55 \cdot 10^6 \text{ m}^2/\text{m}^3$$

$$\text{Specific surface area: } 0.55 \cdot 10^6 \text{ m}^2/\text{m}^3$$

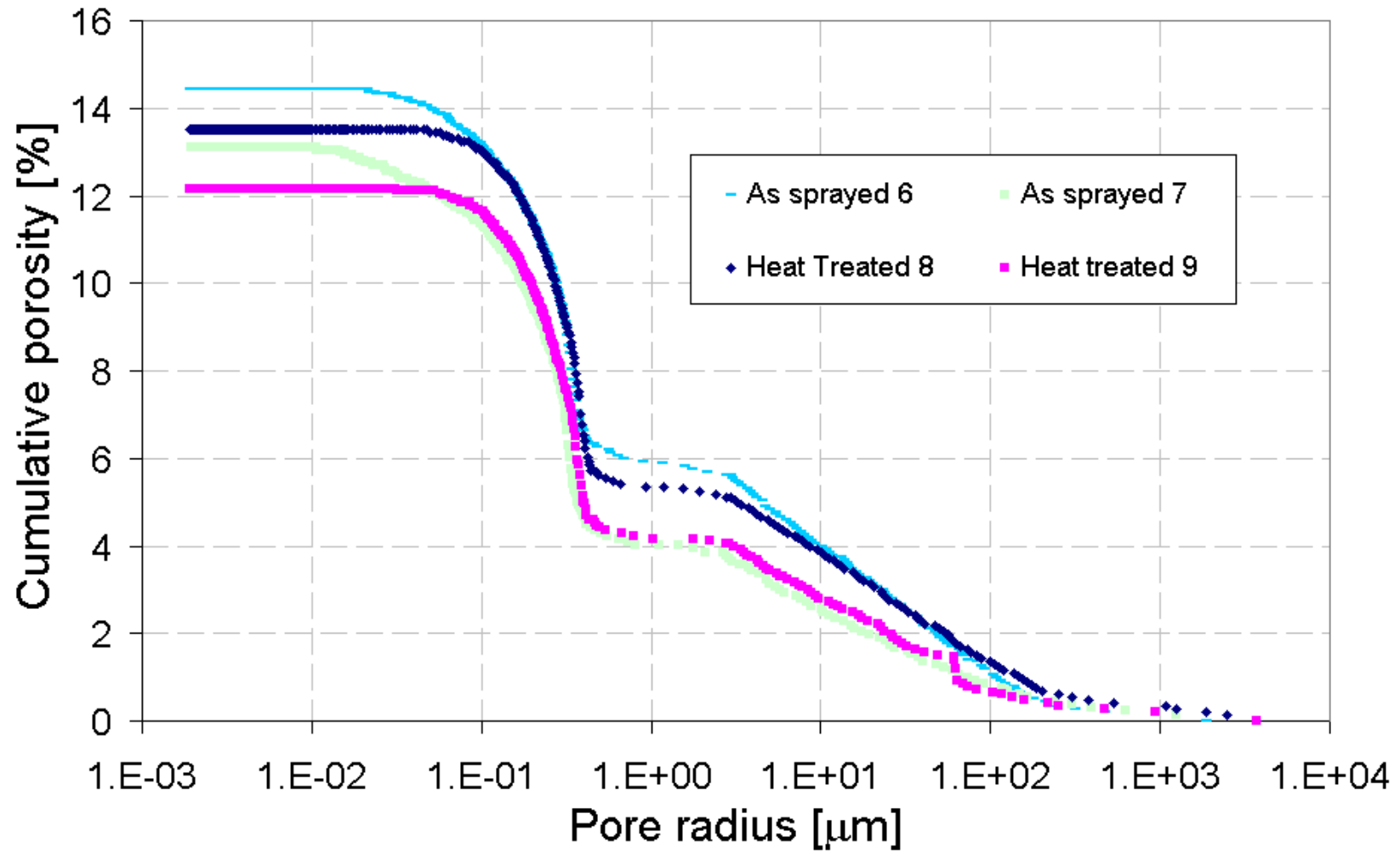
$$\text{Specific surface area: } 0.57 \cdot 10^6 \text{ m}^2/\text{m}^3$$

Porosity 13.8

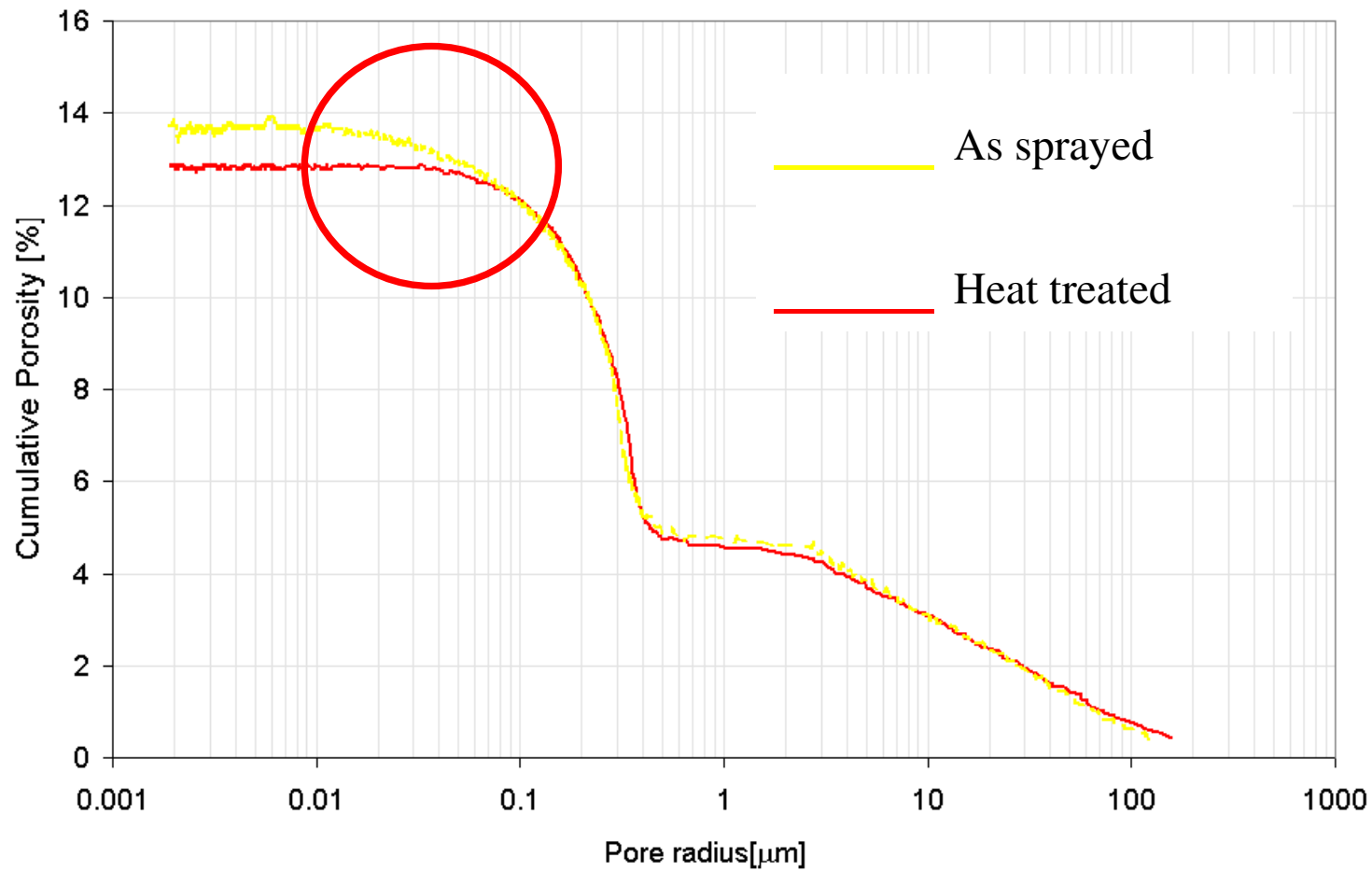
Porosity 12.0

Porosity 10.8

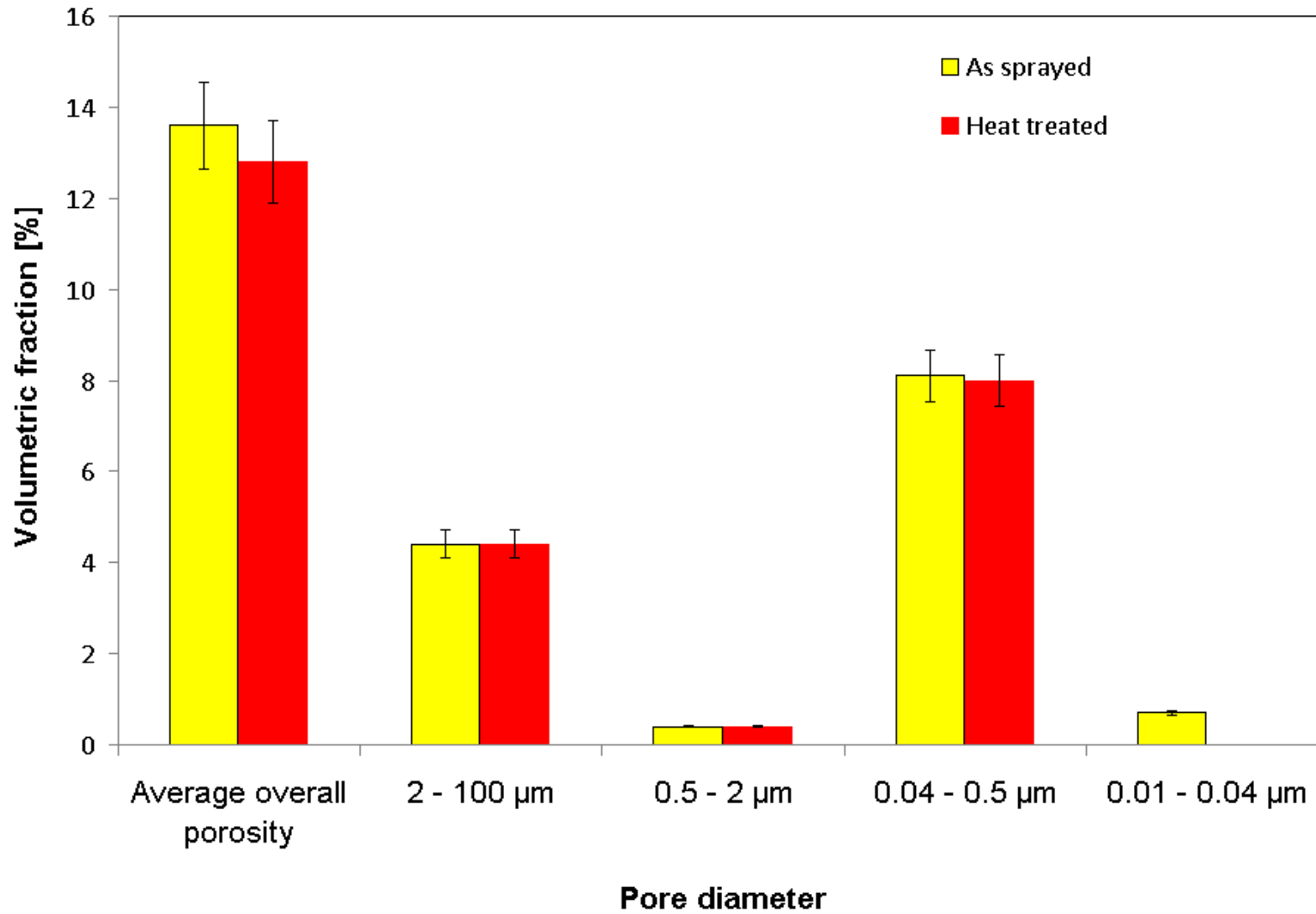
MIP characterisation



MIP characterisation



MIP characterisation



Output of the inversion and sintering codes and MIP

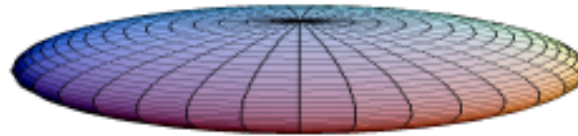


WDS1 heat treated

$$a/c=1/38$$

$$a/c=1/22$$

$$a/c=1/6$$



$$0.088 \pm 0.005 \mu\text{m}$$

$$0.37 \pm 0.05 \mu\text{m}$$

$$0.26 \pm 0.05 \mu\text{m}$$

Specific surface area: $1.82 \cdot 10^6 \text{ m}^2/\text{m}^3$

Specific surface area: $0.39 \cdot 10^6 \text{ m}^2/\text{m}^3$

Specific surface area: $0.58 \cdot 10^6 \text{ m}^2/\text{m}^3$

Specific surface area: $0.70 \cdot 10^6 \text{ m}^2/\text{m}^3$

Porosity 10.1

Porosity 8.8

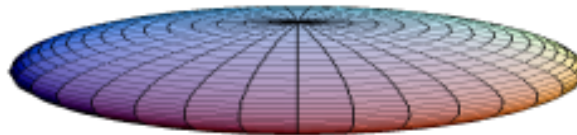
Porosity 13.5

WDS3 heat treated

$$a/c=1/37$$

$$a/c=1/25$$

$$a/c=1/14$$



$$0.163 \pm 0.008 \mu\text{m}$$

$$0.22 \pm 0.05 \mu\text{m}$$

$$0.30 \pm 0.05 \mu\text{m}$$

Specific surface area: $0.55 \cdot 10^6 \text{ m}^2/\text{m}^3$

Specific surface area: $0.55 \cdot 10^6 \text{ m}^2/\text{m}^3$

Specific surface area: $0.57 \cdot 10^6 \text{ m}^2/\text{m}^3$

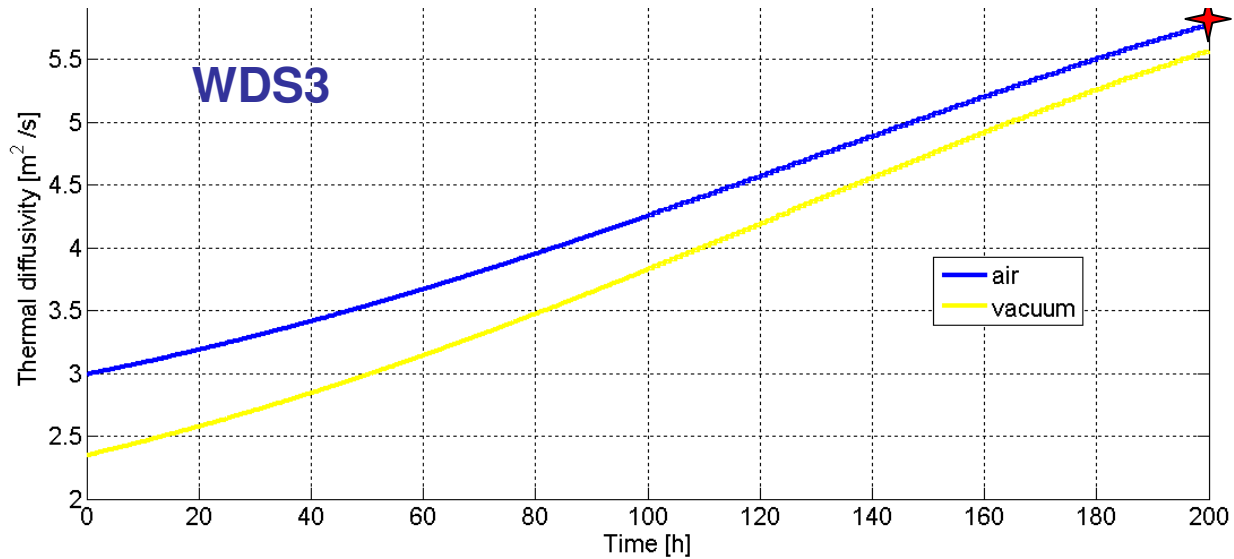
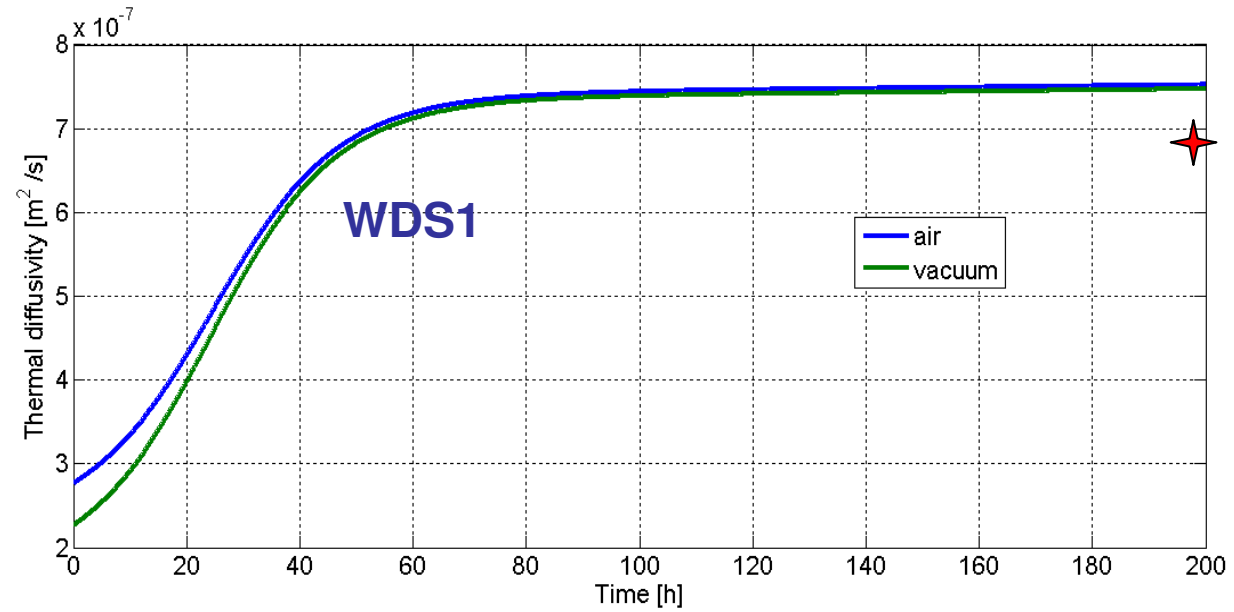
Specific surface area: $0.70 \cdot 10^6 \text{ m}^2/\text{m}^3$

Porosity 12.0

Porosity 10.8

Porosity 12.2

Thermal diffusivity estimations



Conclusive remarks



- ✓ A good agreement between the experimental estimations of **specific surface area** by MIP, Inversion code and Sintering model has been found
 - ✓ The effective **crack opening** estimated by the inversion code is in good agreement with sintering model and IA
 - ✓ **Aspect ratios increase** as estimated by inversion code only partially agrees with that forecasted by the sintering code
 - ✓ **Porosity drop** as measured by MIP resulted smaller than that observed by IA and forecasted by sintering model
 - ✓ **Thermal diffusivity** forecasts in "good" agreement with experimental results for one heat treated sample. A bimodal fine pore distribution could be a possible explanation for the overestimation of thermal diffusivity for sample 1

 - ✓ Thermal diffusivity measurements using different gases and pressures coupled with IA seems to be a useful technique to obtain microstructural parameters to be used as input for the sintering model
-

Through-the-thickness and in-plane thermal diffusivity measurements

Laser thermography

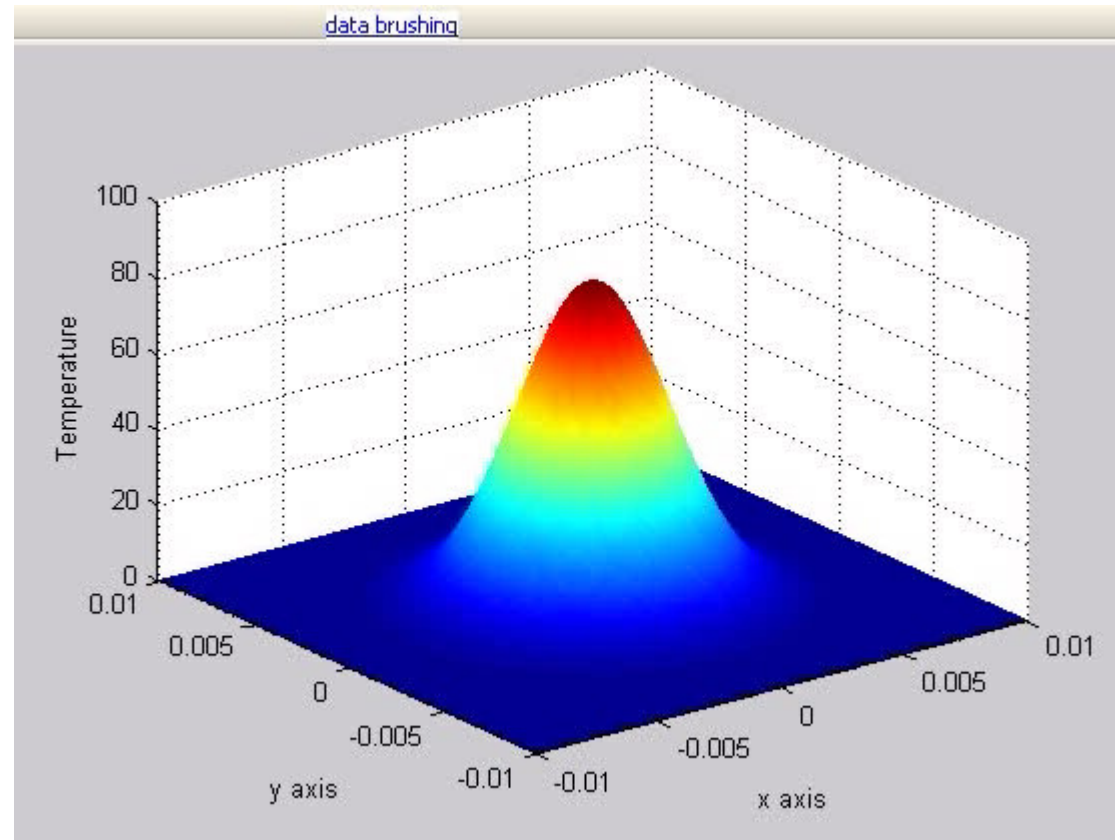


In-plane thermal diffusivity

$$\frac{\Theta(k, z, t)}{\Theta(k = 0, z, t)} = \frac{F(k)}{F(k = 0)} e^{-\alpha_p k^2 t}$$

For slab or semi-infinite body, the spatial Fourier Transform Θ of the in-plane temperature field, decays exponentially in time once normalized by its continuous component. Hence, taking the logarithm of the ratio, it is possible to obtain the in-plane diffusivity from the slope of the fitting straight line.

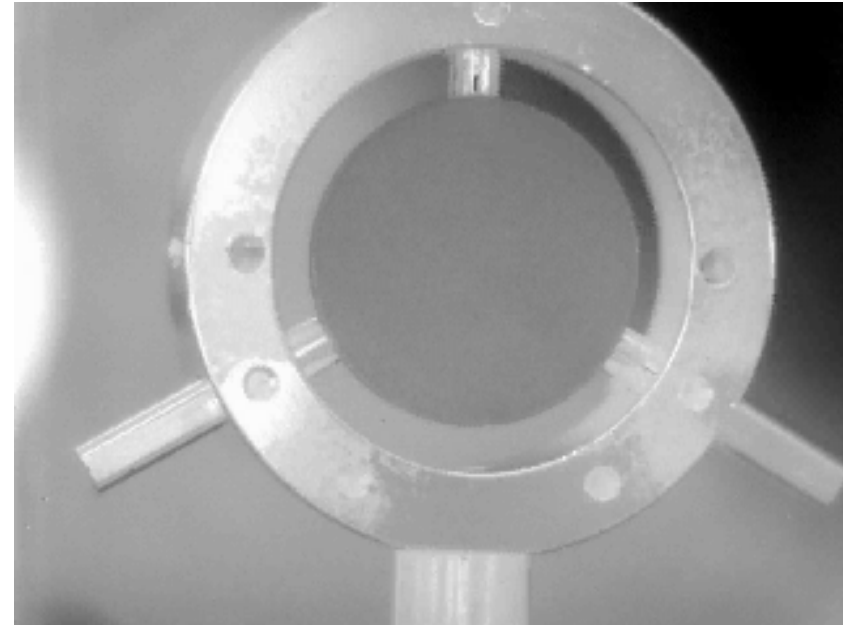
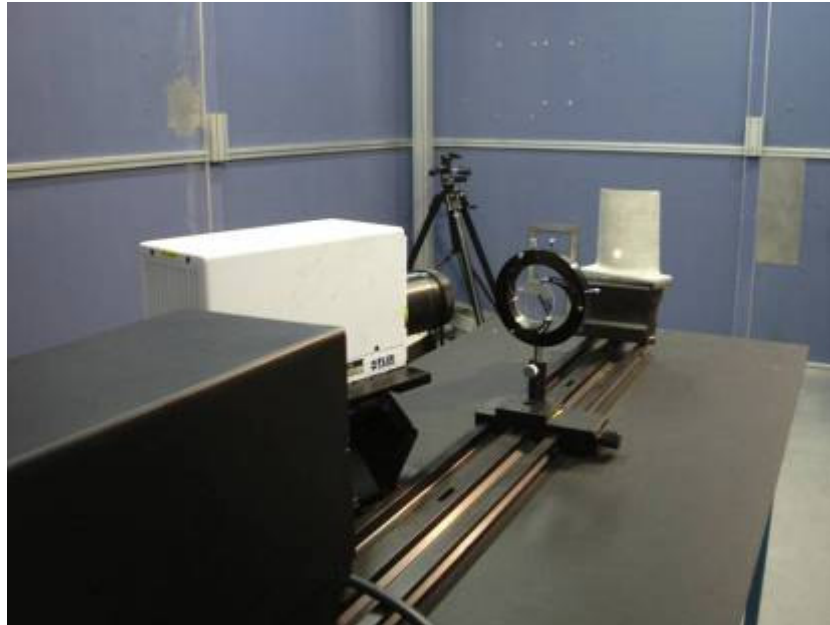
Laser thermography



F. Cernuschi, A. Russo, L. Lorenzoni, A. Figari, *Review of Scientific Instruments* Vol. 72 (10), 2001, p. 1-8.

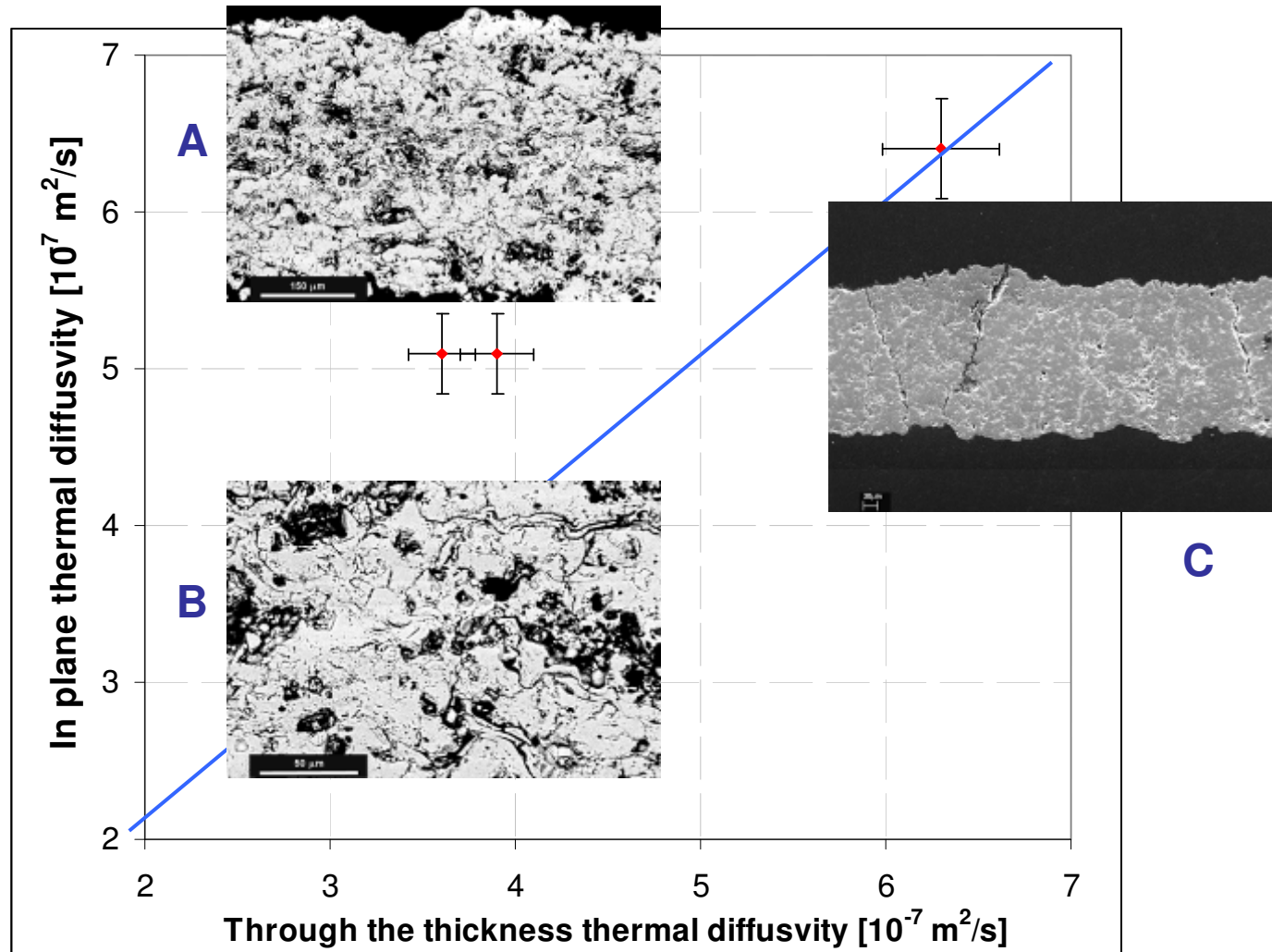
P. Bison, F. Cernuschi, S. Capelli, *Surf. Coat Technol.* [Vol. 205, Issue 10](#), 2011, Pag..3128-3133.

Laser thermography



P.G. Bison, F. Cernuschi, E. Grinzato, S. Marinetti, D. Robba, *Infrared Physics and Technology*, 49 (2007) 286.

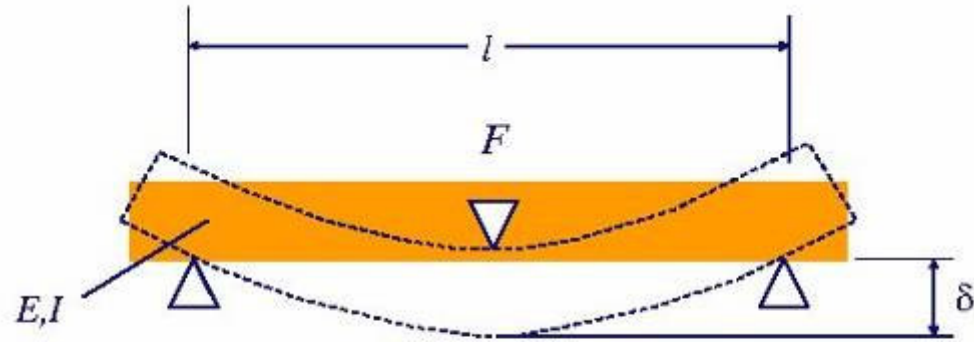
Laser thermography - in-plane thermal diffusivity



Laser thermography - in-plane thermal diffusivity



Elastic modulus measured by three point bending test



$$\sigma = \frac{\pm Fl}{4I} y \quad \delta = \frac{Fl^3}{48EI}$$

Sample	In-depth thermal diffusivity [10 ⁻⁷ m ² s ⁻¹]	In depth thermal conductivity [W/ mK]	In-plane thermal diffusivity [10 ⁻⁷ m ² s ⁻¹]	In-plane thermal conductivity [W/ mK]	In-plane elastic modulus E [GPa]	Simulated in-depth Thermal diffusivity [10 ⁻⁷ m ² s ⁻¹]
A	3.6±0.3	0.8	5.1±0.4	1.1	9.5±0.8	2.8
B	3.9±0.4	0.9	5.1±0.4	1.2	11.4±1.5	3.9
C	6.3±0.1	1.6	6.4±0.2	1.6	1.8±1.5	6.7

Laser thermography - in-plane thermal diffusivity



In the literature several correlations between mechanical and thermo-physical properties of heterogeneous solids have been proposed. It is worth comparing in our case the *in-plane* thermal diffusivity and the *in-plane* elastic modulus of the three samples.

$$\alpha_{pA} / \alpha_{pB} \cong E_A / E_B$$

$$\alpha_{pA} / \alpha_{pC} \cong \alpha_{pB} / \alpha_{pC} \ll E_A / E_C \cong E_B / E_C$$

The assumptions underneath the cross correlation models usually the porosity is supposed to be some orders of magnitude smaller than the typical TBC thickness. In the case of vertical cracks crossing most of TBC thickness this is not true anymore.

Thank you for your kind
attention!